

## PROJECT ADMINISTRATION DATA SHEET



ORIGINAL



REVISION NO. \_\_\_\_\_

Project No. A-4050GTRI/~~OM~~DATE 1 / 8 /85Project Director: J. A. Bruder~~School~~/Lab RAILSponsor: Massachusetts Institute of TechnologyLincoln LaboratoryType Agreement: Purchase Order CX-6211 (Under Gov't Prime F19628-85-C-0002)Award Period: From 11/20/84 To 12/31/84\* (Performance) 12/31/84 (Reports)

Sponsor Amount:

This Change 2-13-85

Total to Date

Estimated: \$ 40,027\$ 40,027Funded: \$ 40,027\$ 40,027Cost Sharing Amount: \$ NoneCost Sharing No: N/ATitle: MMWave Radar Study

## ADMINISTRATIVE DATA

OCA Contact Brian J. Lindberg X4820

1) Sponsor Technical Contact:

2) Sponsor Admin/Contractual Matters:

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Lexington, Massachusetts 02173-0073

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Lexington, Massachusetts 02173-0073

Defense Priority Rating: DO-A7Military Security Classification: N/A(or) Company/Industrial Proprietary: N/A

## RESTRICTIONS

See Attached N/A Supplemental Information Sheet for Additional Requirements.

Travel: Foreign travel must have prior approval – Contact OCA in each case. Domestic travel requires sponsor approval where total will exceed greater of \$500 or 125% of approved proposal budget category.

Equipment: Title vests with none proposed or anticipated

## COMMENTS:

\*Note: No cost extension through 2/15/85 has been requested.



## COPIES TO:

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Library  
Project File  
Other \_\_\_\_\_

SR519 2-N-B

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OFFICE OF CONTRACT ADMINISTRATION

SPONSORED PROJECT TERMINATION/CLOSEOUT SHEET

Date 8/2/85

Project No. A-4050 ~~XXXX~~/Lab RAIL

Includes Subproject No.(s) \_\_\_\_\_

Project Director(s) J. A. Bruder ~~OFFICE~~ / GIT

Sponsor Massachusetts Institute of Technology, Lincoln Laboratory

Title MMWave Radar Study

Effective Completion Date: 2/15/85 (Performance) 2/15/85 (Reports)

Grant/Contract Closeout Actions Remaining:

- ☐ None
- ☒ Final Invoice or Final Fiscal Report
- ☐ Closing Documents
- ☒ Final Report of Inventions
- ☒ Govt. Property Inventory & Related Certificate
- ☐ Classified Material Certificate
- ☐ Other \_\_\_\_\_

Continues Project No. \_\_\_\_\_ Continued by Project No. \_\_\_\_\_

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Jones

BRIEFING MATERIAL

presented at

LINCOLN LABORATORY  
MASSACHUSETTS INSTITUTE OF TECHNOLOGY  
Boston, Massachusetts

October 18, 1984

by

N. C. Currie  
W. A. Holm  
D. L. Odom  
J. A. Scheer  
R. N. Trebits

GEORGIA TECH RESEARCH INSTITUTE  
GEORGIA INSTITUTE OF TECHNOLOGY  
Atlanta, Georgia



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## MIT/LL INSTRUMENTATION SAR SYSTEM

### BRIEFING OUTLINE

EXECUTIVE SUMMARY

RADAR SYSTEM DESIGN CONSIDERATIONS

RF/IF HARDWARE REALIZATION

DATA ACQUISITION & SYSTEM CONTROL(AIRBORNE)

DATA PROCESSING, GROUND BASED

MEASUREMENT & CALIBRATION

PROGRAM ASPECTS





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MIT/LL - DARPA

## INSTRUMENTATION SAR PROGRAM

**GTRI** HAS DEVELOPED A STRAIGHTFORWARD, LOW RISK, COMPLIANT PROGRAM IN RESPONSE TO MIT/LL - DARPA NEED FOR AN AIRBORNE INSTRUMENTATION SAR DATA COLLECTION SYSTEM.



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## EXPERIENCE BASE

TEST AND INSTRUMENTATION

EXTENSIVE EXPERIENCE  
(>25 RECENT PROGRAMS)

AIRBORNE

VARIOUS PROGRAMS USING  
GTRI AND CONTRACTOR AIRCRAFT

POLARIZATION

RECOGNIZED AUTHORITY IN  
POLARIZATION RELATED PROCESSING

MILLIMETER WAVE

30 YEARS EXPERIENCE IN  
MILLIMETER WAVE R&D

HIGH RESOLUTION PROCESSING

PROVEN HIGH RANGE RESOLUTION  
TECHNIQUES

STRAIGHT FORWARD, LOW RISK SAR  
APPROACH

ALL AIRBORNE DATA RECORDED  
(NO PRE PROCESSING)

GTRI STAFF AUGMENTED



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## TEST ENVIRONMENT

RANGE:	SHORT(1500M)
RANGE SWATH:	SMALL(150M)
PROCESSING:	NON. REAL TIME POLAR PROCESSING NOT REQ'D
FLIGHT ENVIRONMENT:	BENIGN, STRAIGHT & LEVEL
SYNTHETIC ARRAY TIME:	SHORT $T = \frac{R\lambda}{2\ell V} (\cong 0.25\text{sec})$
UNCOMPENSATED ACCELERATION ALLOWABLE:	LARGE $a < \frac{\ell^2 V^2}{R^2 \lambda} (\cong 0.04\text{M/sec}^2)$



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## OUTLINE OF APPROACH

### PULSED Ka - BAND RADAR:

70 WATT TWTA

CHIRPED/COMPRESSED PULSE

STEPPED FREQUENCY FOR ULTIMATE RANGE RESOLUTION

POLARIZATION DIVERSE ON TRANSMIT

DUAL POLARIZED ON RECEIVE

} FULL POLARIZATION MATRIX

### SAR - STRIPMAP, SPOTLIGHT

ZERO DOPPLER PROCESSING

RECORD RADAR, NAV, MOCOMP DATA ON TAPE

DATA PROCESSING VAN FOR

- QUICK-LOOK PROCESSING
- PRODUCTION PROCESSING

### RAR -

FORWARD LOOKING, SCANNING REAL BEAM



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## SYSTEM ENHANCING FEATURES

IMPROVED CROSSPOLARIZATION  
ISOLATION:

NOT ACHIEVABLE W/O  
PENCIL (CIRCULAR) BEAM  
WHICH COMPROMISES  
PERFORMANCE

3 - AXIS POSITIONER:

COST -

INCREASED RANGE SWATH:

HIGHER AVERAGE DATA RATE  
DRIVES RECORDER COST

INCREASED CROSSRANGE  
SWATH:

NO EFFECT IN STRIPMAP  
LARGER QUANTUM ANGULAR  
STEPS IN SPOTLIGHT

REDUCED CROSSRANGE  
RESOLUTION (DBS):

NOT ADVISED



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## RADAR SYSTEM DESIGN CONSIDERATIONS

1. RANGE RESOLUTION APPROACH

2. SAR

- ISSUES
- STRIP MAP MODE
- SPOTLIGHT MODE

3. RAR



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## RANGE RESOLUTION ISSUES

<u>TECHNIQUE</u>	<u>COMMENTS</u>
1. REAL PULSE	PEAK POWER LIMITED DATA ACQUISITION SAMPLING PROBLEM
2. FMCW OR INTERRUPTED FWCW	SEVERE LINEARITY REQUIREMENTS
3. CHIRP COMPRESSION	DATA ACQUISITION SAMPLING PROBLEM
4. PULSE-TO-PULSE FREQUENCY STEPPING	SYNTHESIZER SWITCHING SPEED



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## RANGE RESOLUTION APPROACH

- PULSE-TO-PULSE FREQUENCY STEPPING
  - COMSTRON FREQUENCY SYNTHESIZER
  - DESIRED RANGE RESOLUTION

PROBLEM: SNR MARGINAL

SOLUTION: PRECEED FREQUENCY STEPPING WITH CHIRP PULSE COMPRESSION

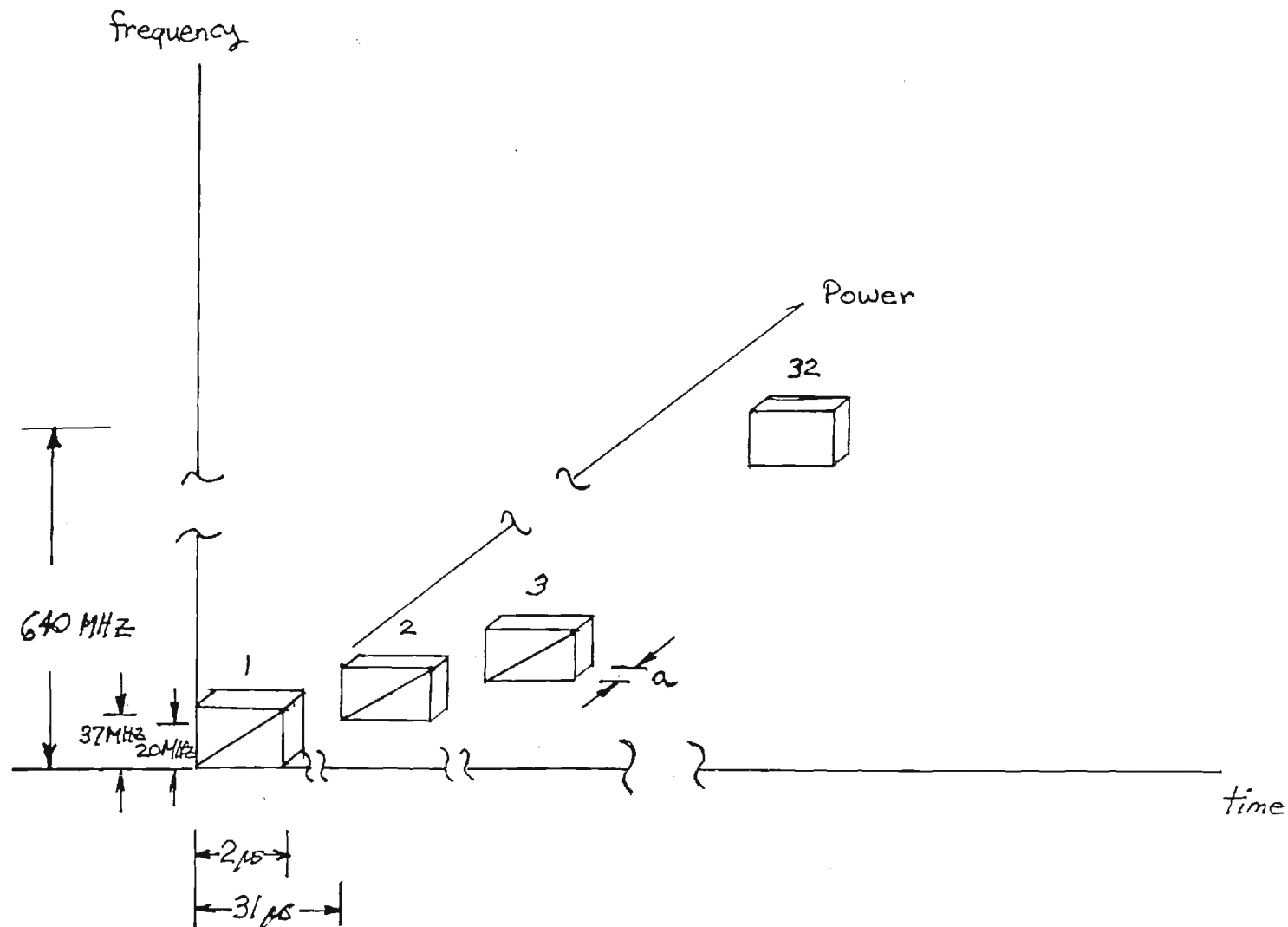
( 17 dB ADDITIONAL PROCESSING GAIN )





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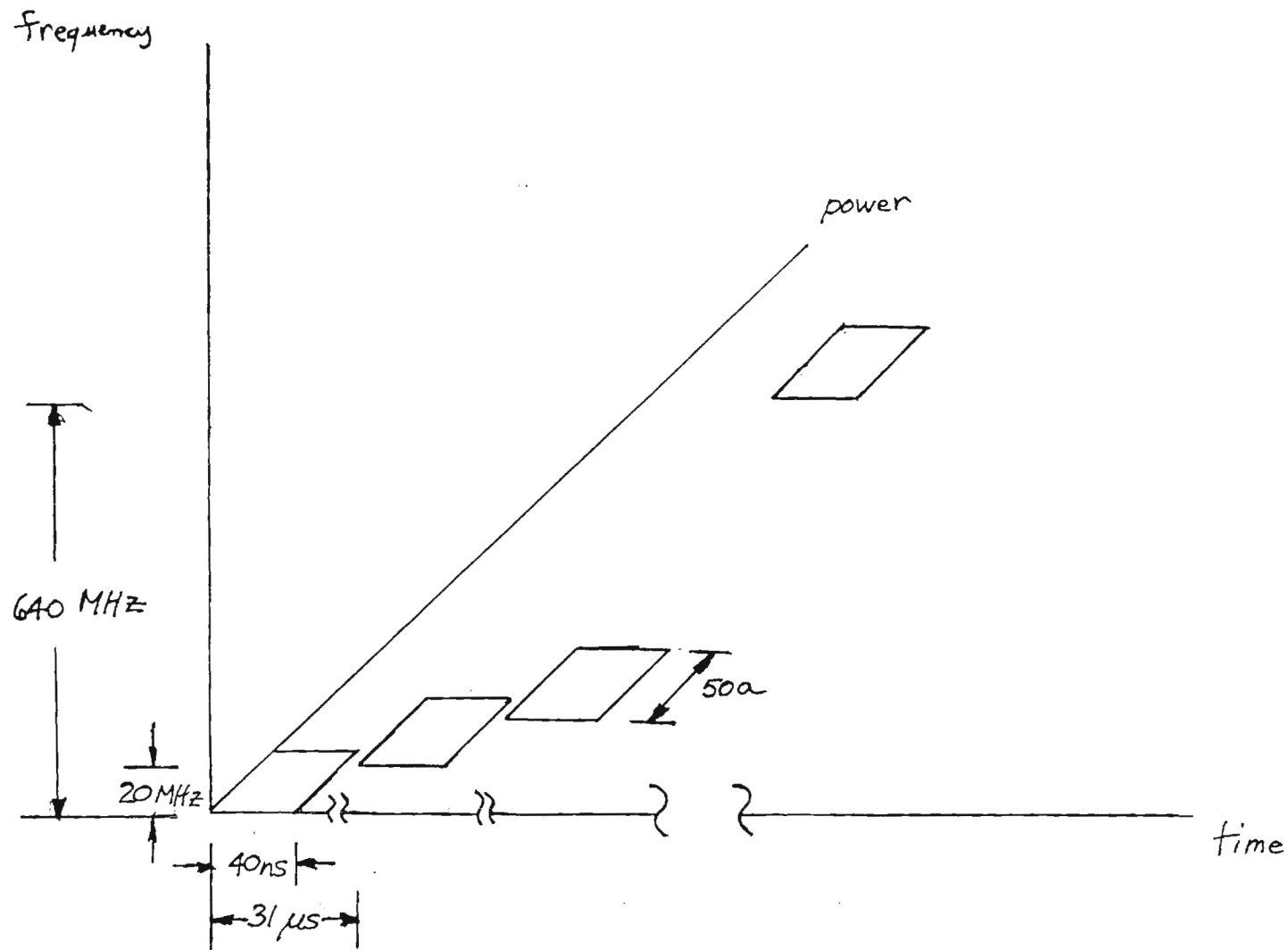
## RANGE RESOLUTION APPROACH (RECEIVED SIGNAL)





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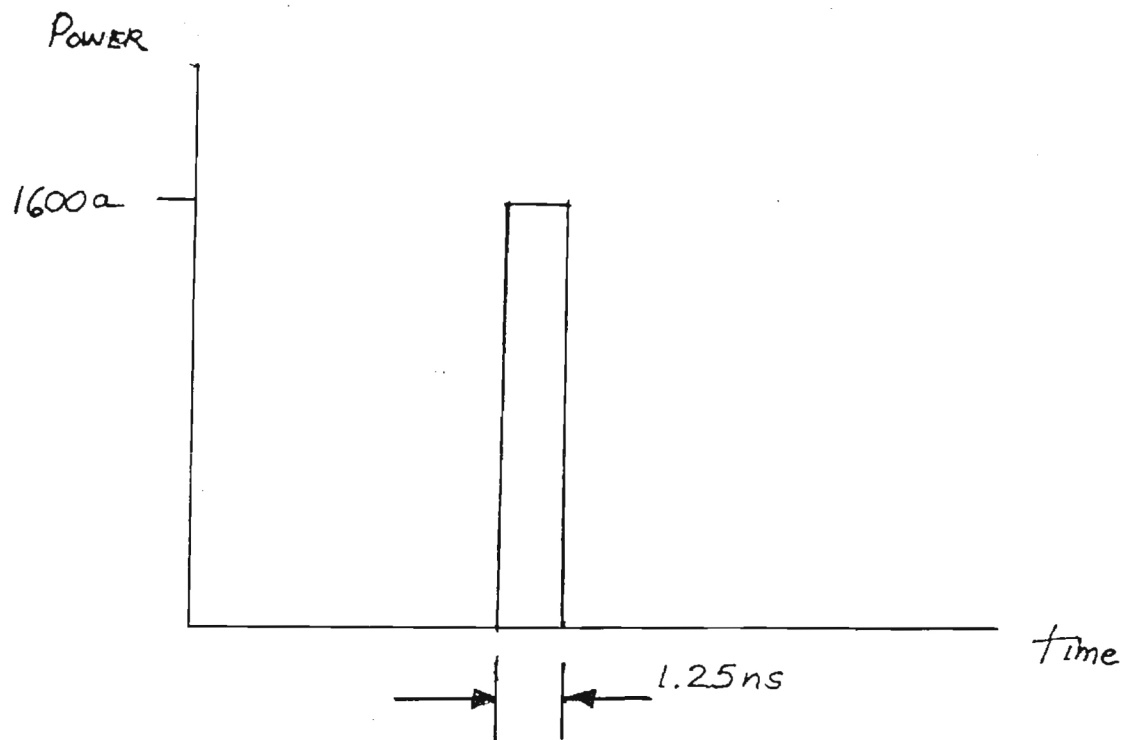
## RANGE RESOLUTION APPROACH (AFTER SAW)





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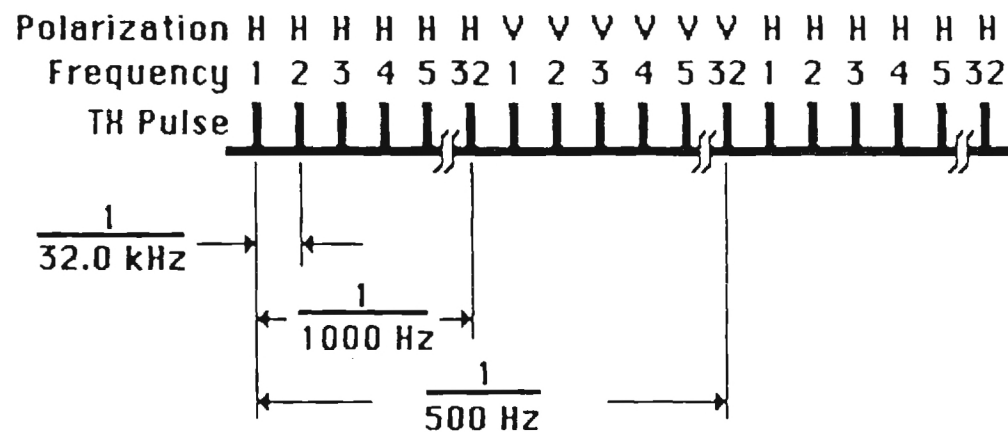
## RANGE RESOLUTION APPROACH (AFTER RANGE FFT)





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## RADAR SYSTEM TIMING DIAGRAM





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## GENERAL SAR ISSUES

- HIGH X-RANGE RESOLUTION
- LOW INTEGRATED SIDELobe LEVEL
  - MOTION COMPENSATION
  - RANGE WALK
- MOTION COMPENSATION
- PROCESSING SPEED



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## PROGRAM SAR ISSUES

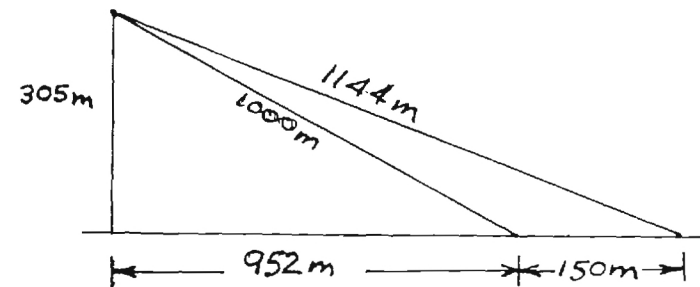
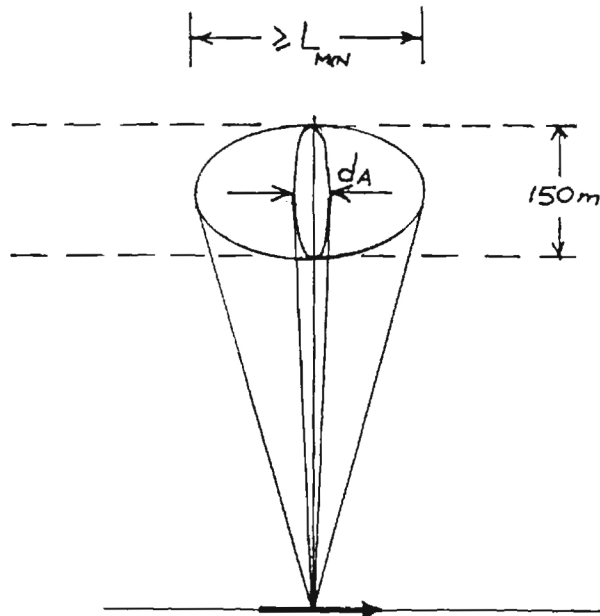
### TEST MEASUREMENT CONDITIONS

- STRAIGHT, LEVEL, CONSTANT SPEED A/C MOTION (NO HIGH-G MANUVER)  
(HELPS MoCOMP COMPLEXITY)
  
- SHORT RANGES
  - SHORT APERTURE TIMES (REDUCES ISL)
  
  - NO "RANGE WALK" CORRECTION (REDUCES PROCESSOR COMPLEXITY, ISL)
  
- NO REAL-TIME PROCESSING REQUIREMENTS AT FULL RESOLUTIONS



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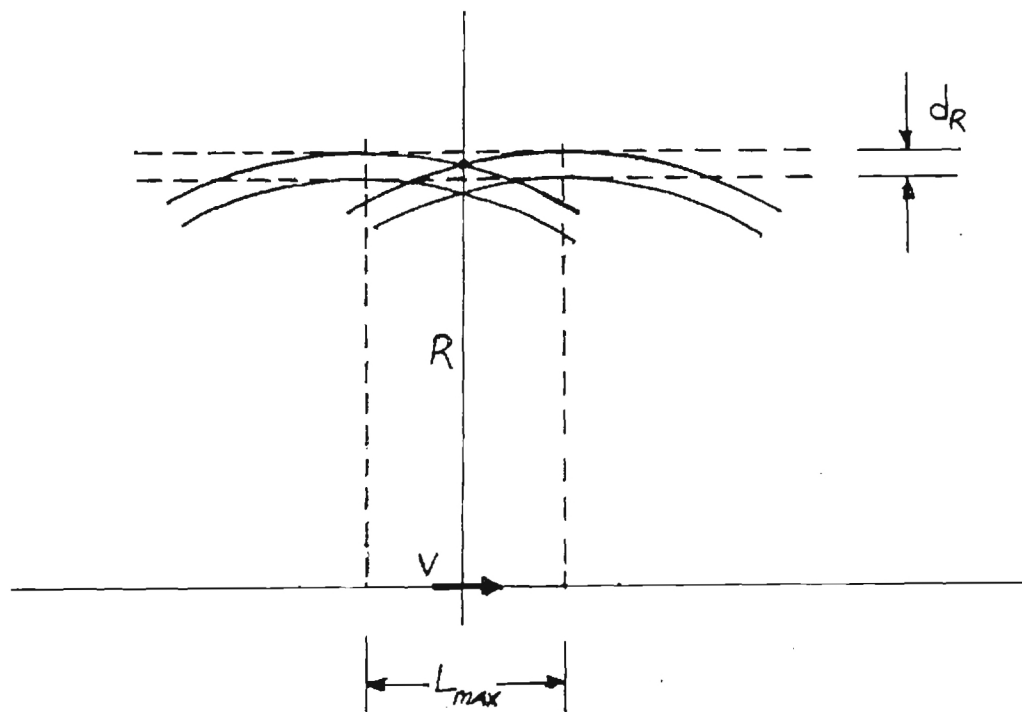
## SAR STRIP-MAP MODE





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## RANGE WALK (STRIP MAP MODE)



$$L_{max} = 2\sqrt{Rd_R}$$





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## SYNTHETIC APERTURE LENGTH (STRIP MAP MODE)

$$d_A > \frac{\lambda R}{2L} \rightarrow L_{\min} = \frac{\lambda R}{2d_A} \quad (\text{TO ACHIEVE X-RANGE RESOLUTION REQUIREMENTS})$$

$$L < 2\sqrt{Rd_R} \rightarrow L_{\max} = 2\sqrt{Rd_R} \quad (\text{TO PREVENT "RANGE WALK"})$$

LET  $R_{\max}$  BE MAXIMUM RANGE BEFORE RANGE WALK PROCESSING IS REQUIRED

$$L_{\min} = L_{\max} \rightarrow R_{\max} = 6.2 \text{ km}$$



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## SYNTHETIC APERTURE LENGTH (STRIP MAP MODE)

$$L_{\min} = 50 \text{ ft}$$

$$R = 1085 \text{ m}$$

$$L_{\max} = 119 \text{ ft}$$

$$L = L_{\min} = 50 \text{ ft}$$

∴ RANGE WALK IS NOT A PROBLEM IN STRIP MAP MODE



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## SAR ANTENNA X-RANGE DIMENSION (STRIP MAP MODE)

$$\frac{R\lambda}{\ell} > L \quad \rightarrow \quad \ell_{\max} = \frac{R\lambda}{L}$$

(TO ILLUMINATE LARGE ENOUGH PATCH ON GROUND  
FOR X-RANGE RESOLUTION)

$$\frac{2V}{\ell} \leq (f_r)_e \quad \rightarrow \quad \ell_{\min} = \frac{2V}{(f_r)_e}$$

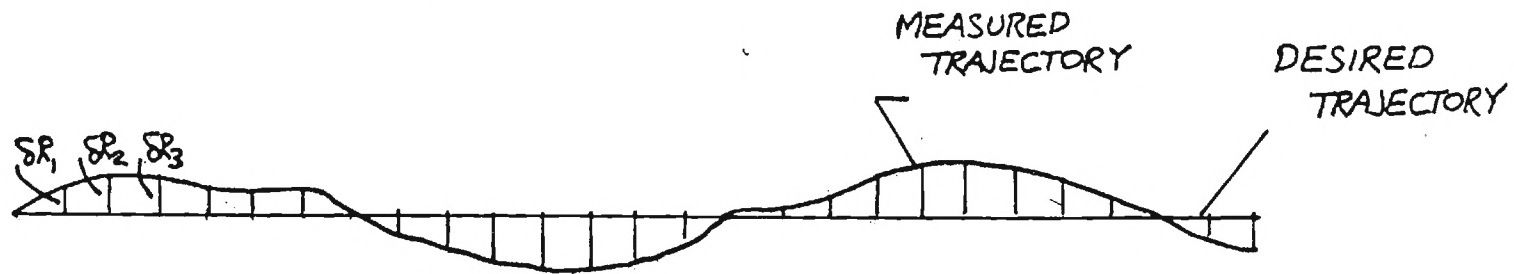
(TO PREVENT DOPPLER AMBIGUITIES)

$$\ell_{\max} = 24 \text{ inches}$$



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## MOTION COMPENSATION (STRIP MAP MODE)



$\{SR_x\}$  - Line of sight range differences between measured and desired trajectories.



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## MOTION COMPENSATION (STRIP-MAP MODE)

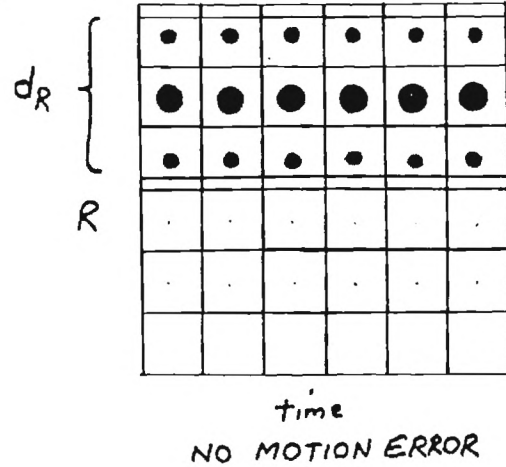
- PHASE CORRECTIONS  $\Delta\phi_R = 2\pi\delta R_R/\lambda$
- RANGE BIN REGISTRATION & COMPLEX INTERPOLATION



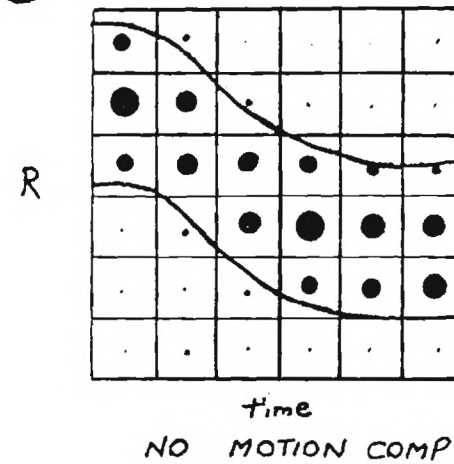
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## RANGE BIN REGISTRATION AND INTERPOLATION

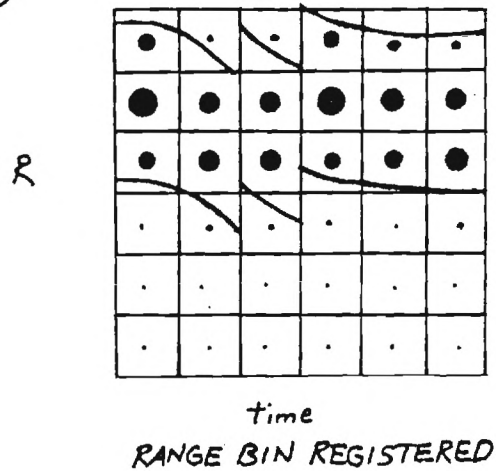
①



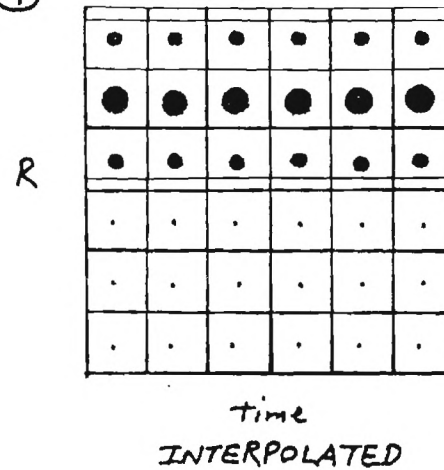
②



③



④





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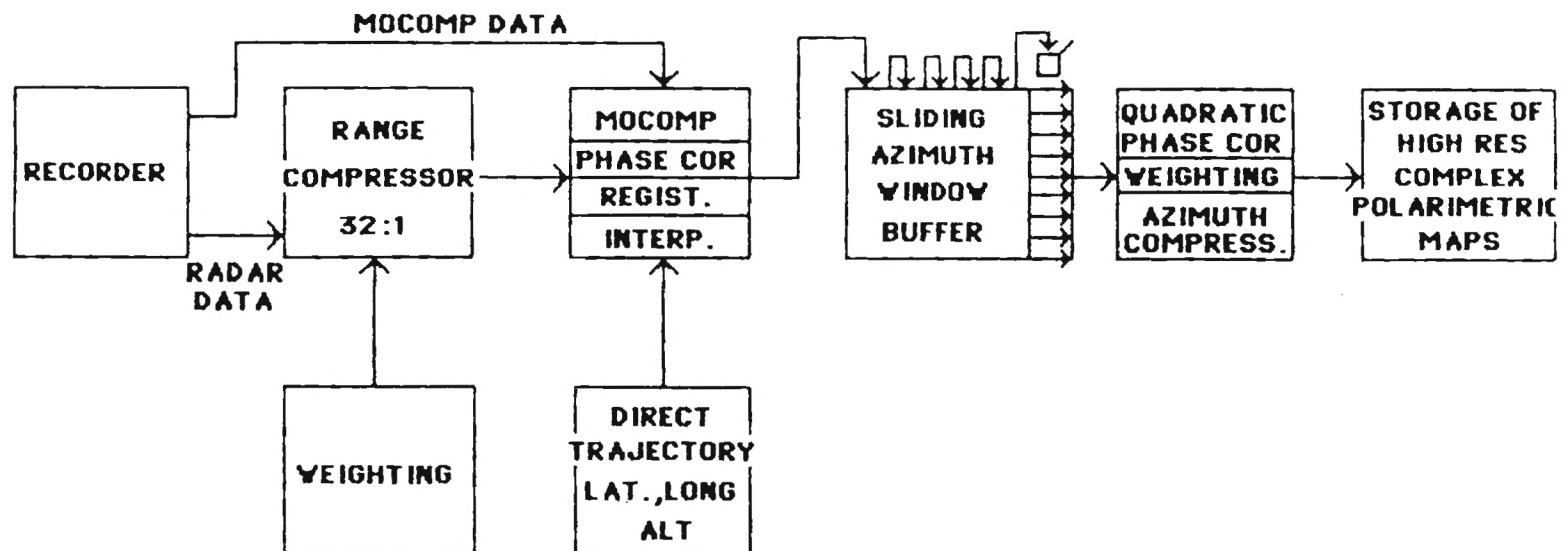
## PROCESSING SPEED

- NO REAL-TIME IMAGING REQUIREMENT
  
- THEREFORE, PERFORM "SLIDING WINDOW" AZIMUTHAL COMPRESSION
  - NEW RANGE SWEEP IN, OLDEST RANGE SWEEP OUT OF BUFFER
  
  - COMPLEX SUM IN AZIMUTH
  
  - PRODUCES ONE HIGH X-RANGE RESOLUTION RANGE SWEEP



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## SAR GROUND PROCESSING STATION (STRIP MAP MODE ONLY)

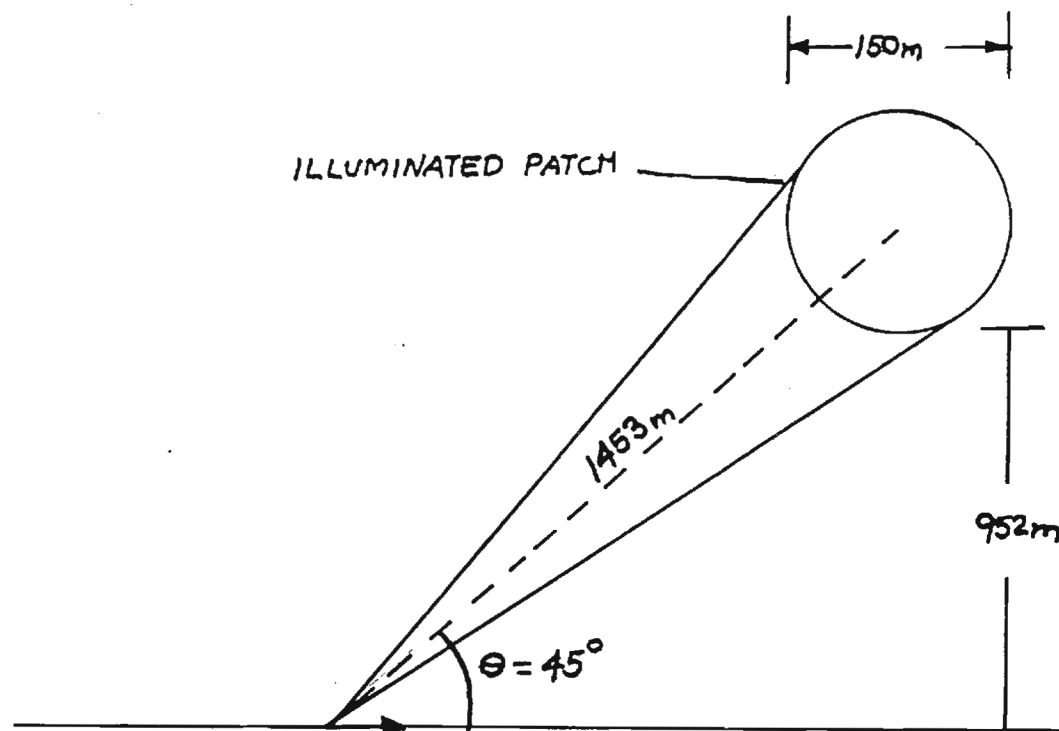






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## SAR CONVENTIONAL SPOTLIGHT MODE



$$-45^\circ \leq \theta \leq 45^\circ$$



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## SYNTHETIC APERTURE LENGTH (CONVENTIONAL SPOTLIGHT MODE)

$$d_A > \frac{\lambda R}{2L} \rightarrow L_{\min} = \frac{\lambda R}{2d_A} \quad (\text{TO ACHIEVE X-RANGE RESOLUTION REQUIREMENTS})$$

$$L < \frac{R d_R}{L_{\min}} \rightarrow L_{\max} = \frac{R d_R}{L_{\min}} = \frac{2d_A d_R}{\lambda} \quad (\text{RANGE WALK})$$

$$L_{\min} = 67 \text{ ft} \quad R = 1453 \text{ m}$$

$$L_{\max} = 71 \text{ ft}$$

∴ "RANGE WALK EFFECTS BECOMING NOTICEABLE DUE TO

1.  $L_{\max} \approx L_{\min}$

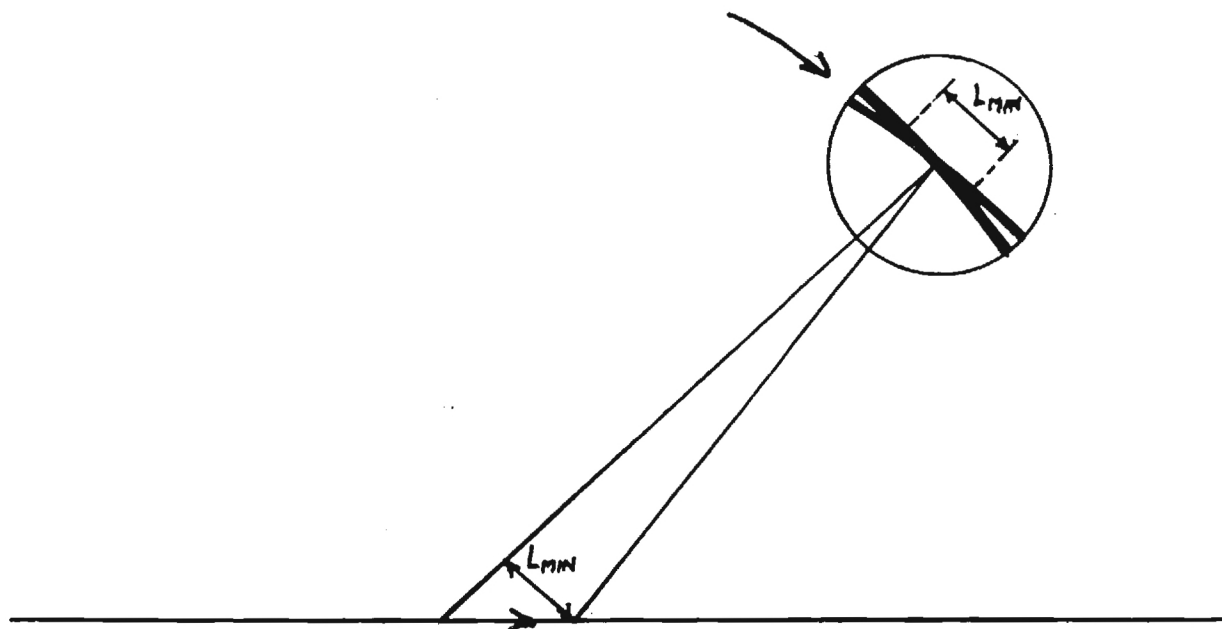
2. UN-DOPPLER PROCESSED ILLUMINATED PATCH INCREASES ISL



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## RANGE WALK (CONVENTIONAL SPOTLIGHT MODE)

ILLUMINATED PATCH





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## SAR ANTENNA X-RANGE DIMENSION (SPOTLIGHT MODE)

$$\frac{R\lambda}{\ell} > W \rightarrow \ell_{\max} = \frac{R\lambda}{W}$$

(TO KEEP ENTIRE SPOTLIGHT AREA  
ILLUMINATED)

$$\frac{2V}{\ell} < (f_r)_e \rightarrow \ell_{\min} = \frac{2V}{(f_r)_e}$$

(TO PREVENT DOPPLER AMBIGUITIES)

$$\ell_{\max} = 2.4 \text{ inches}$$

$$\text{IF } \ell_{\min} = \ell_{\max} \rightarrow f_{re} = 2.1 \text{ kHz (3.2 kHz 50\% margin)}$$

$$f_r = 136.5 \text{ kHz (205 kHz 50\% margin)}$$

REQUIRED PRF TOO HIGH



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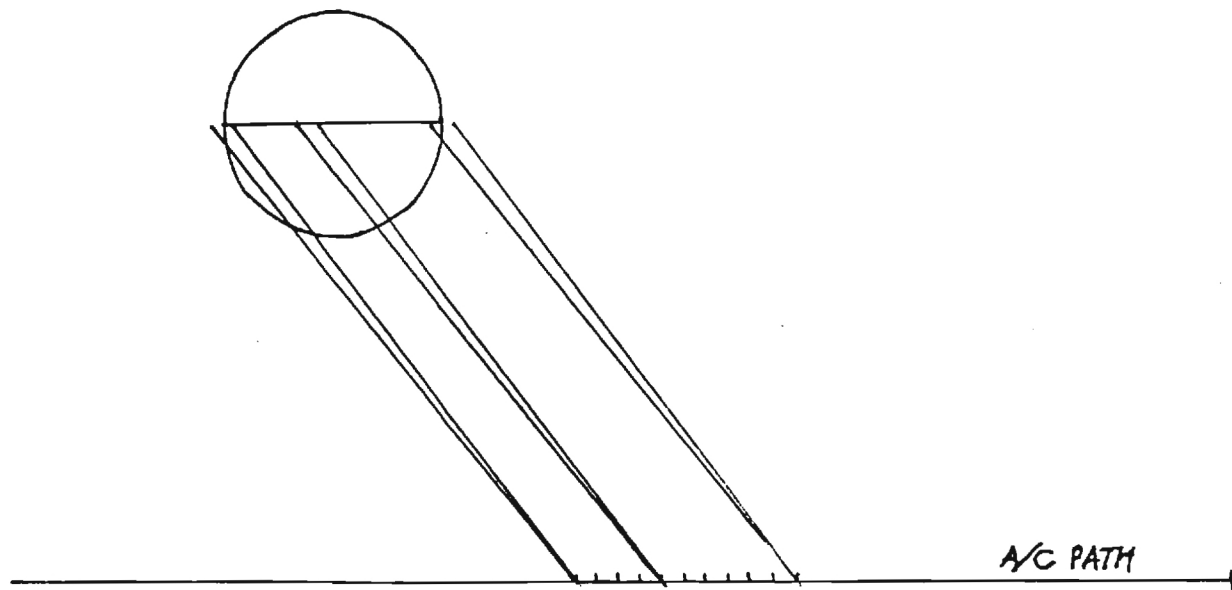
## MODIFIED SPOTLIGHT MODE (SQUINT STRIP-MAP/SPOTLIGHT)

1. FIXED-ANGLE SQUINT STRIP-MAP AS REAL BEAM TRAVERSES SPOTLIGHT PATCH (SAME REAL BEAM GROUND ILLUMINATION AS IN STRIP-MAP MODE);
2. BACK-SCAN ANTENNA IN AZIMUTH TO PUT BEAM ON LEADING EDGE OF PATCH;
3. REPEAT (1) AND (2) UNTIL SQUINT ANGLE GOES FROM  $-45^{\circ}$  TO  $45^{\circ}$ .



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## MODIFIED SPOTLIGHT MODE





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## MODIFIED SPOTLIGHT MODE ADVANTAGES

- NO RANGE-WALK PROBLEM (BACK TO STRIP MAP MODE)
- NO HIGH PRF REQUIREMENT (ILLUMINATING SMALL PATCH ON GROUND AS IN STRIP MAP MODE)
- MINIMUM DIFFERENCES BETWEEN STRIP-MAP AND SPOTLIGHT MODES :

ANTENNA IS STEPPED-SCANNED IN AZIMUTH  
(EVERY 2-3 SECONDS)

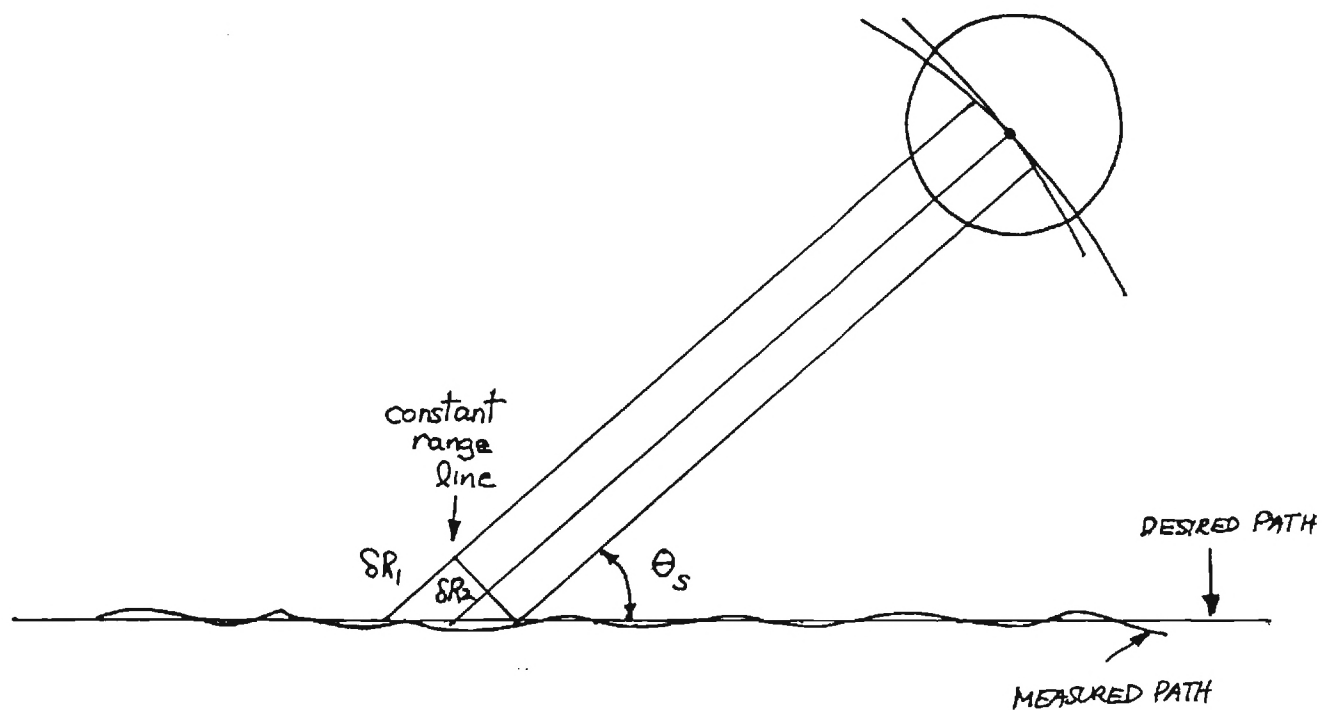
ADDITIONAL RANGE REGISTRATION/INTERPOLATION

QUADRATIC PHASE FACTOR IS FUNCTION OF SQUINT ANGLE



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## ADDITION RANGE CORRECTIONS REQUIRED WITH MODIFIED SPOTLIGHT MODE

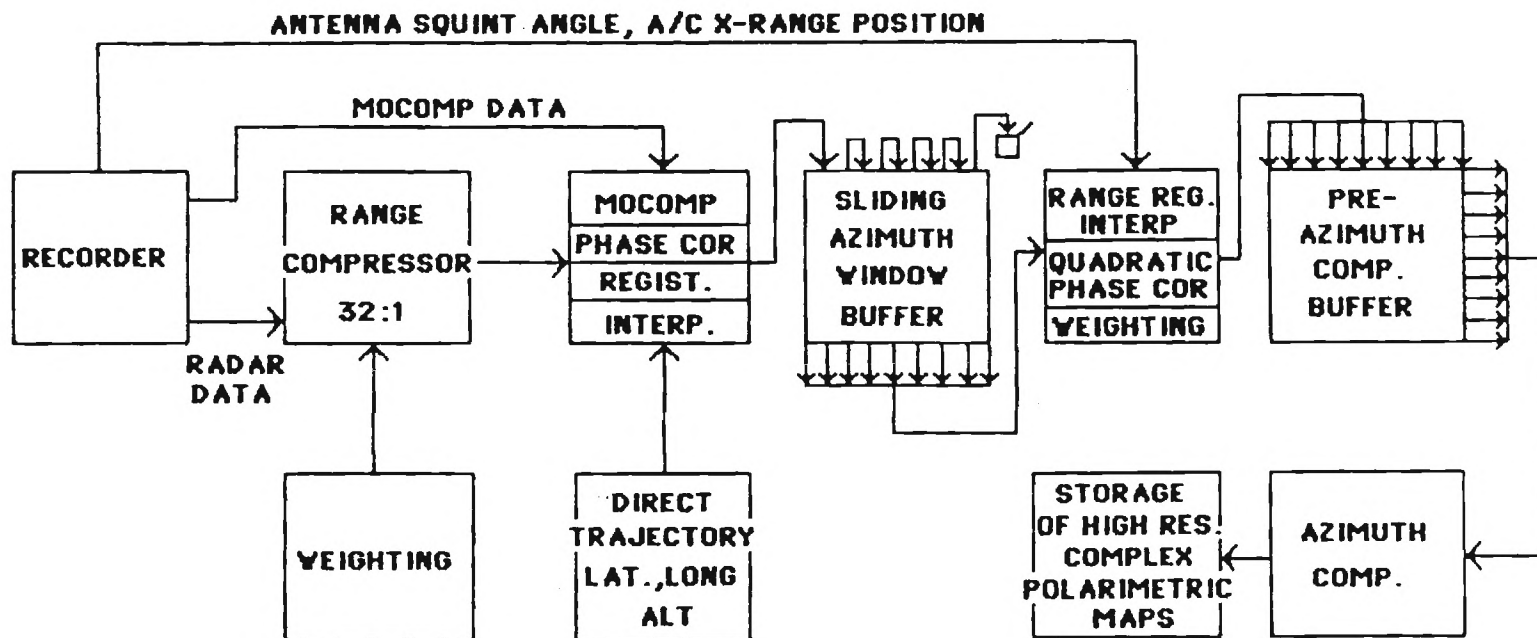






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## SAR GROUND PROCESSING STATION (STRIP MAP & SPOTLIGHT MODES)





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## RAR APPROACH

Antenna:

1.5° El  
3° Az

Scan:

±30° @ 1 Hz

Range Coverage:

150 M @ 1500 M up to 30°  
(90 M @ 60°)

Overlap:

65 M/s Velocity

Az Resolution

40 M @ 1500 M

PRF:

500 Hz Effective

Range Resolution:

Same as SAR



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## RADAR SYSTEM SALIENT FEATURES

COHERENT, FREQUENCY STEPPED, CHIRPED/COMPRESSED WAVEFORM

### TRANSMITTER:

FREQUENCY: 33.2 - 33.84 GHz

PEAK POWER: 70 WATTS

PRF: 32 KHz

PULSE: 2  $\mu$ sec, 37 MHz LINEAR CHIRP

FREQUENCY STEP: 32, 20 MHz STEPS



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## RADAR SYSTEM SALIENT FEATURES

### RECEIVER:

S.A.W. COMPRESSION:	50:1 TO 40 nsec
A-TO-D RATE:	32 MHz
NUMBER OF BINS:	32
FFT PROCESS/BIN:	32 POINT
NUMBER OF SYNTHETIC BINS:	1024
SYNTHETIC PULSE LENGTH:	$\approx 2$ nsec
EFFECTIVE PRF:	500 Hz/POLARIZATION



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## RADAR SYSTEM SALIENT FEATURES

ANTENNA: HORN - FED LENS (SAR, RAR)

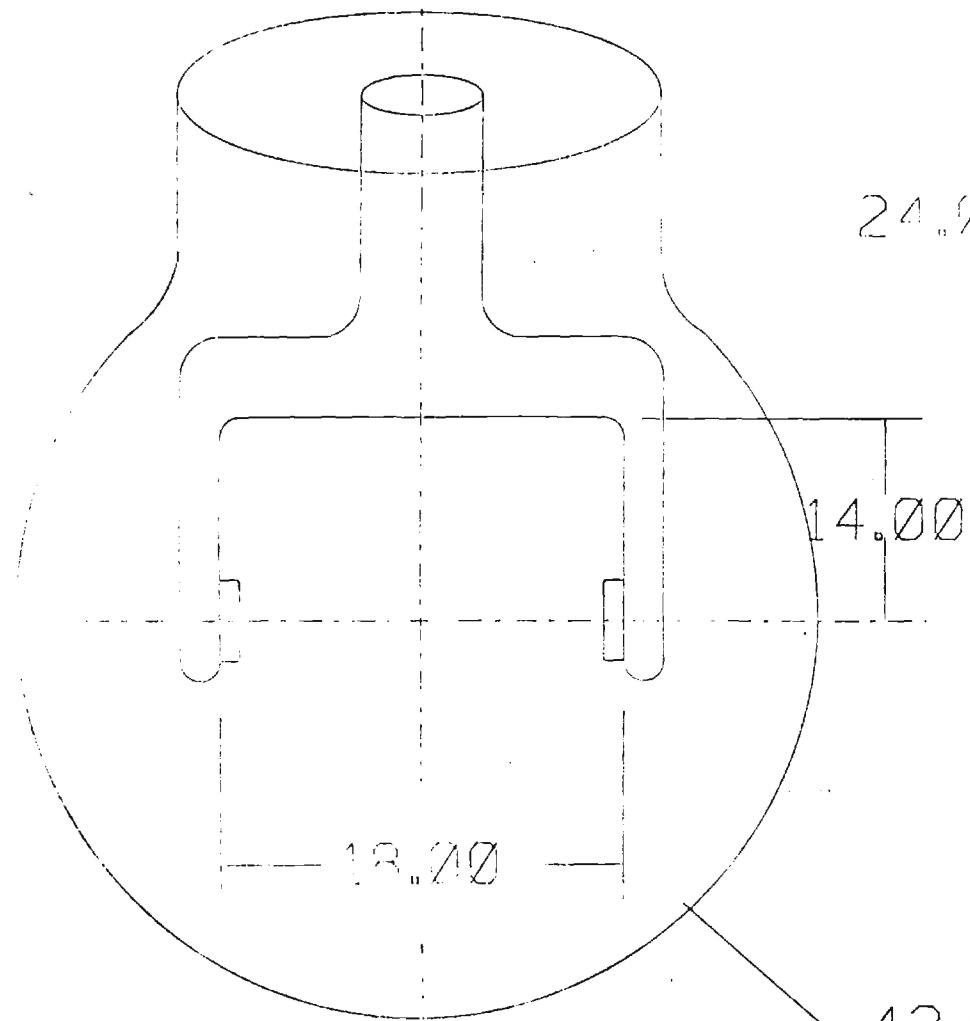
AZ:  $1.5^{\circ}$  (.43M)

EL:  $3.0^{\circ}$  (.22M)

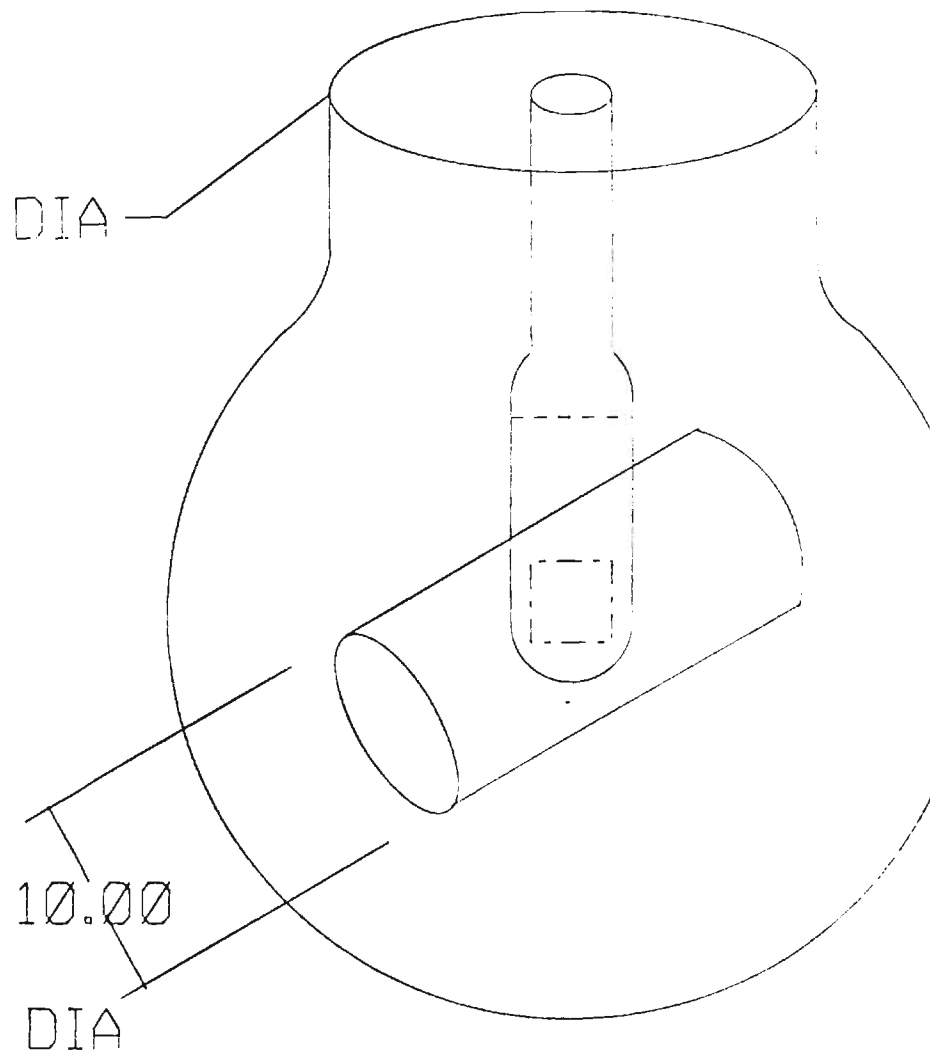
GAIN: 38 dB

SIDELOBES: 30 dB PEAK

POLARIZATION: 25 dB ISOLATION, 1.0 dB  
ELLIPTICITY



24.00 DIA

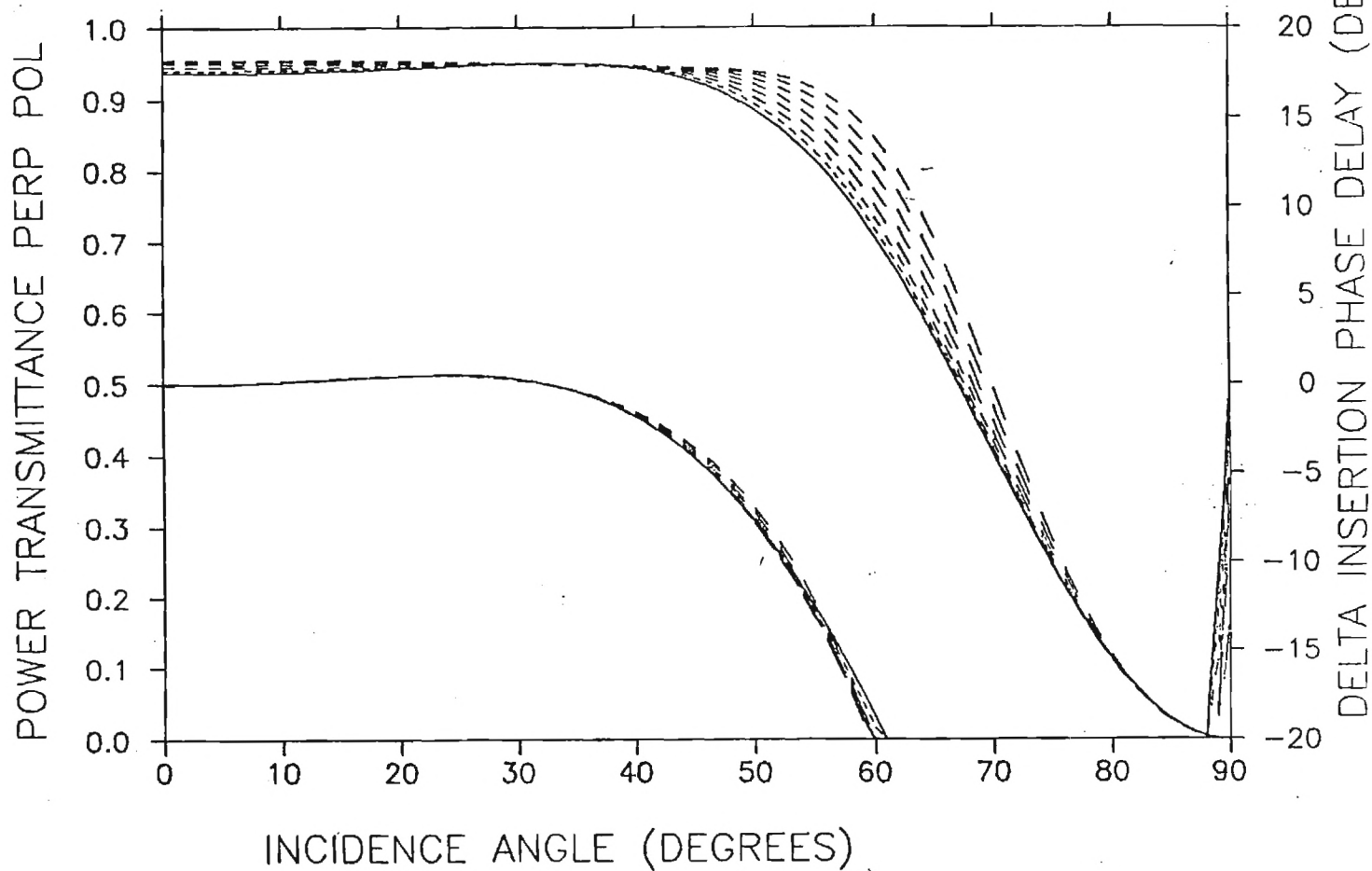


10.00

42.00 DIA

CORE

LEGEND

0.0300 IN.0.0400 IN.0.0500 IN.0.0600 IN.0.0700 IN.0.0800 IN.0.0900 IN.0.1000 IN.

A-SANDWICH (3 LAYER) RADOME WALL - F/34.0 - 2 Half-wave Skins

POWER TRANSMITTANCE PARALLEL POL

CORE

LEGEND

0.0300 IN.

0.0400 IN.

0.0500 IN.

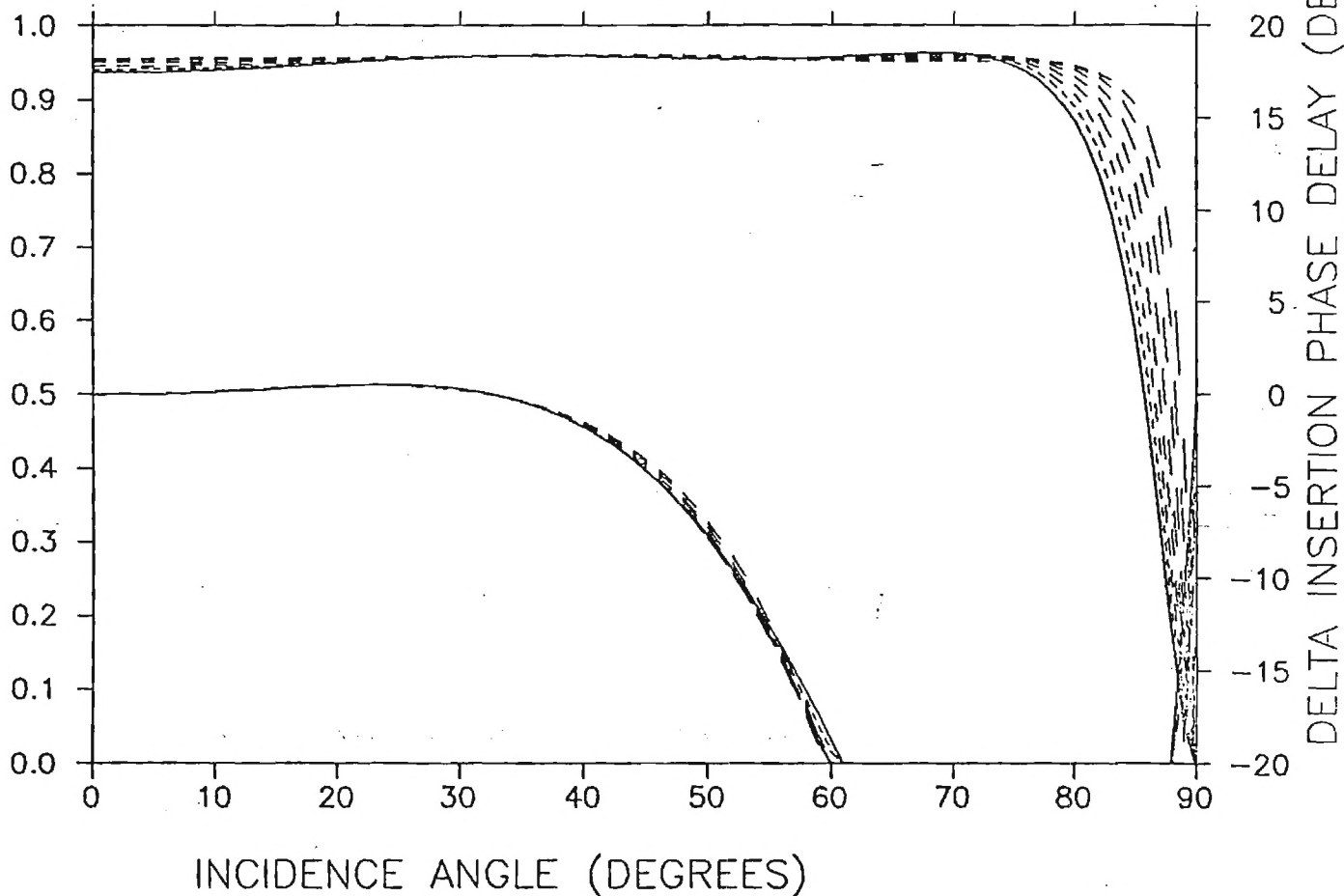
0.0600 IN.

0.0700 IN.

0.0800 IN.

0.0900 IN.

0.1000 IN.

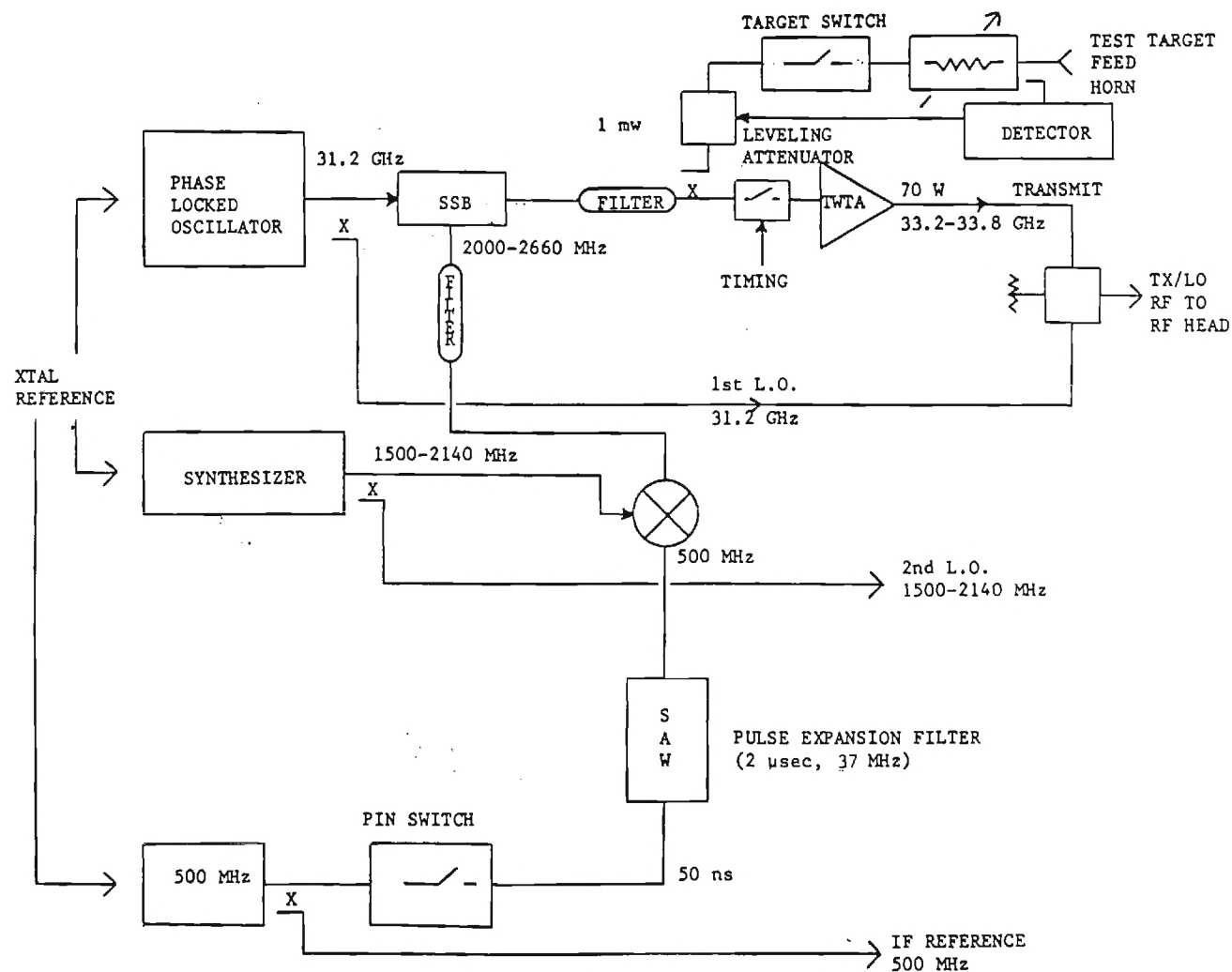


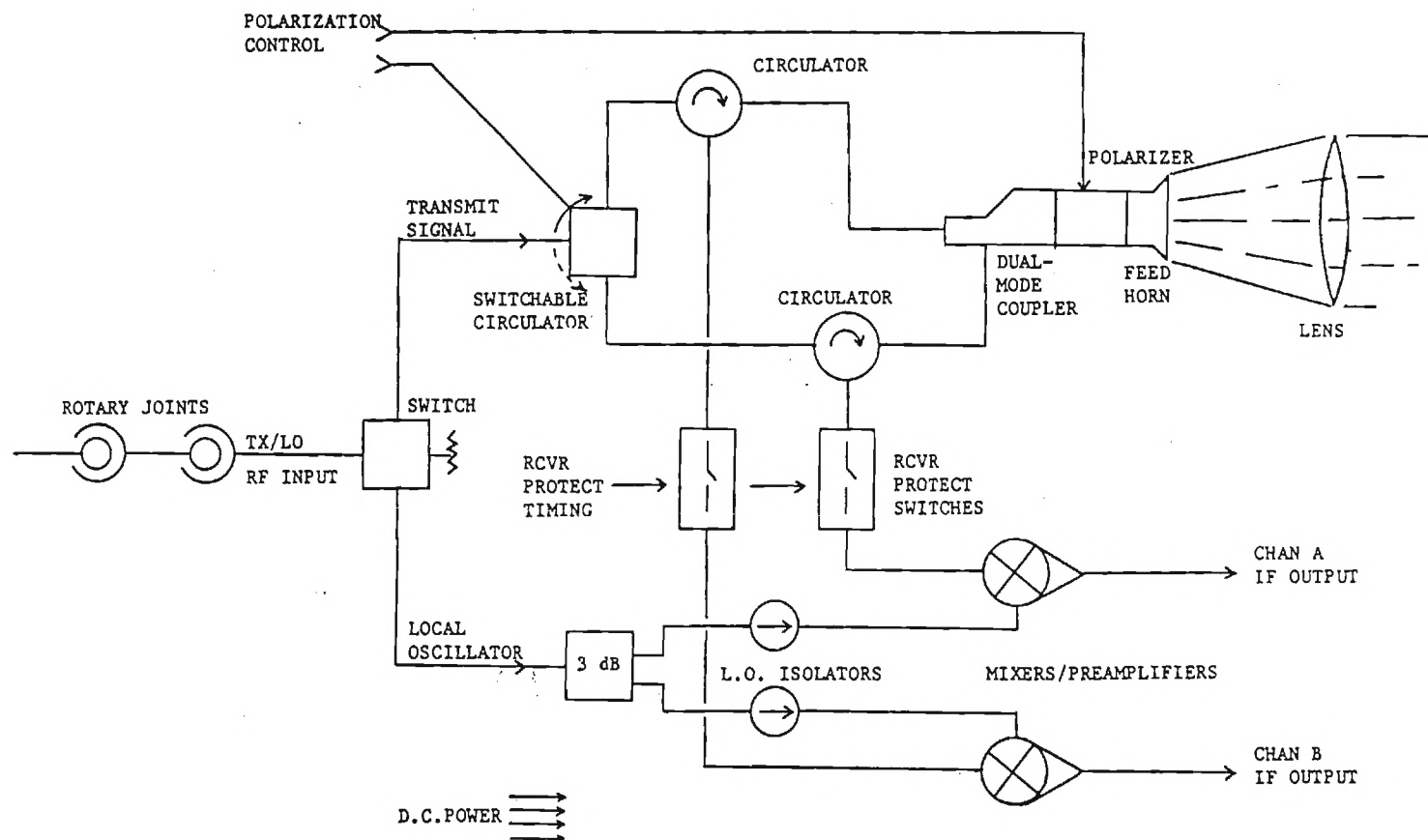




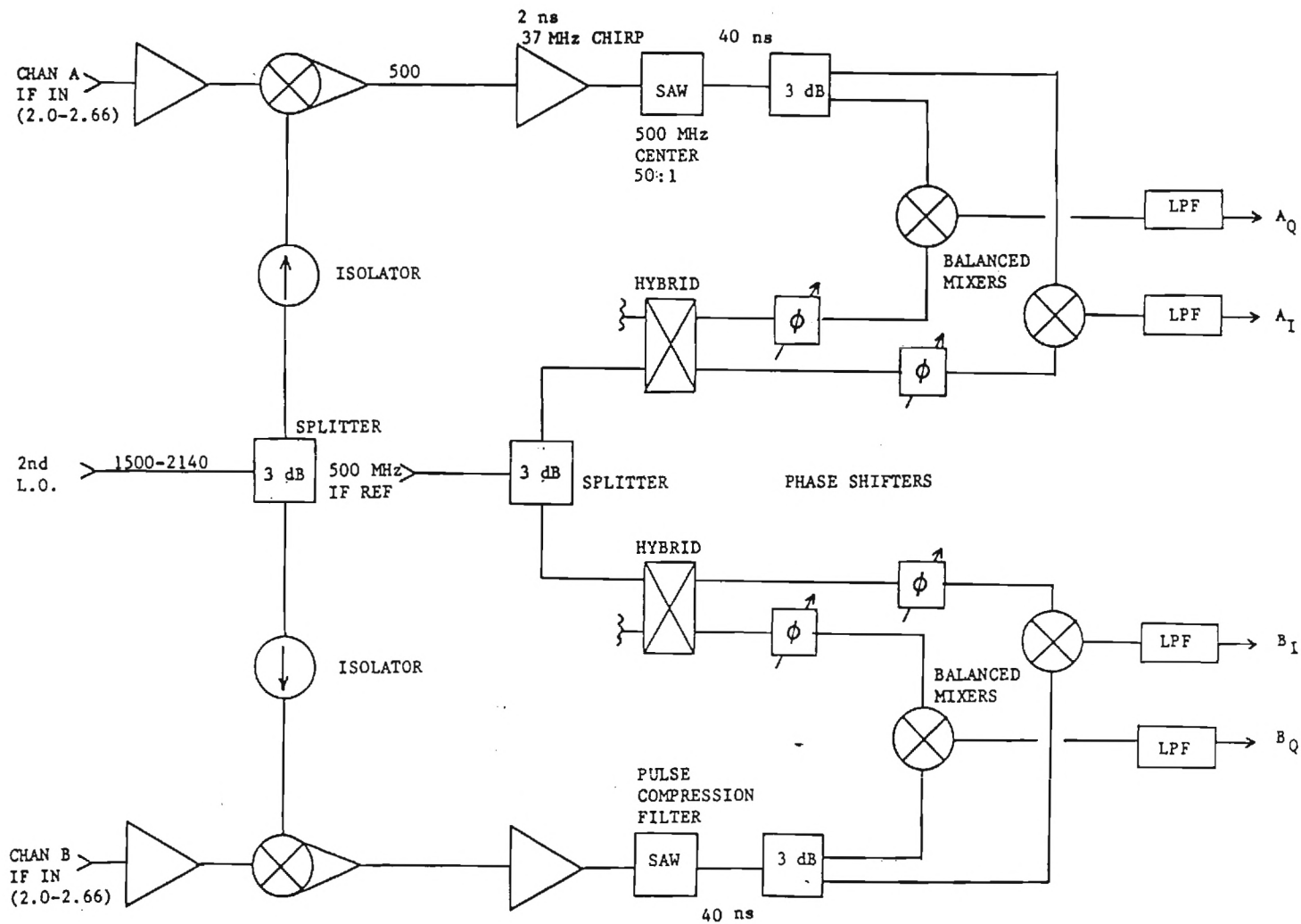
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## RF SUBSYSTEM



**RF HEAD**

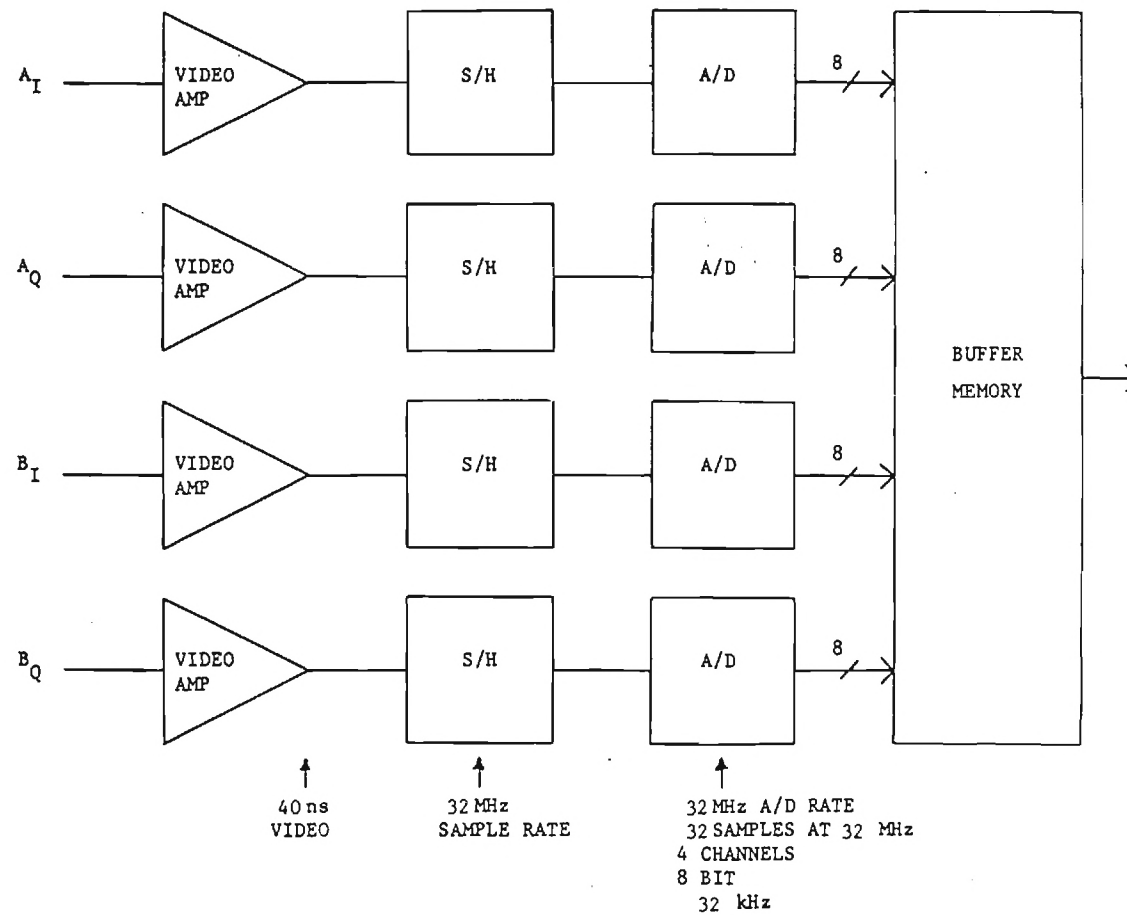
## IF SUBSYSTEM





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## RADAR DATA ACQUISITION SUBSYSTEM





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ADVANCED DETECTION TECHNOLOGY PROGRAM

AIRBORNE DATA ACQUISITION AND SYSTEM CONTROL



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## OBJECTIVES

1. Record radar reflectivity, motion compensation data, time codes, status flags, etc.
2. Provide range gate delay to the analog-to-digital converters.
3. Control antenna pointing direction.
4. Provide built-in test capability.
5. Provide in-flight assessment of data quality



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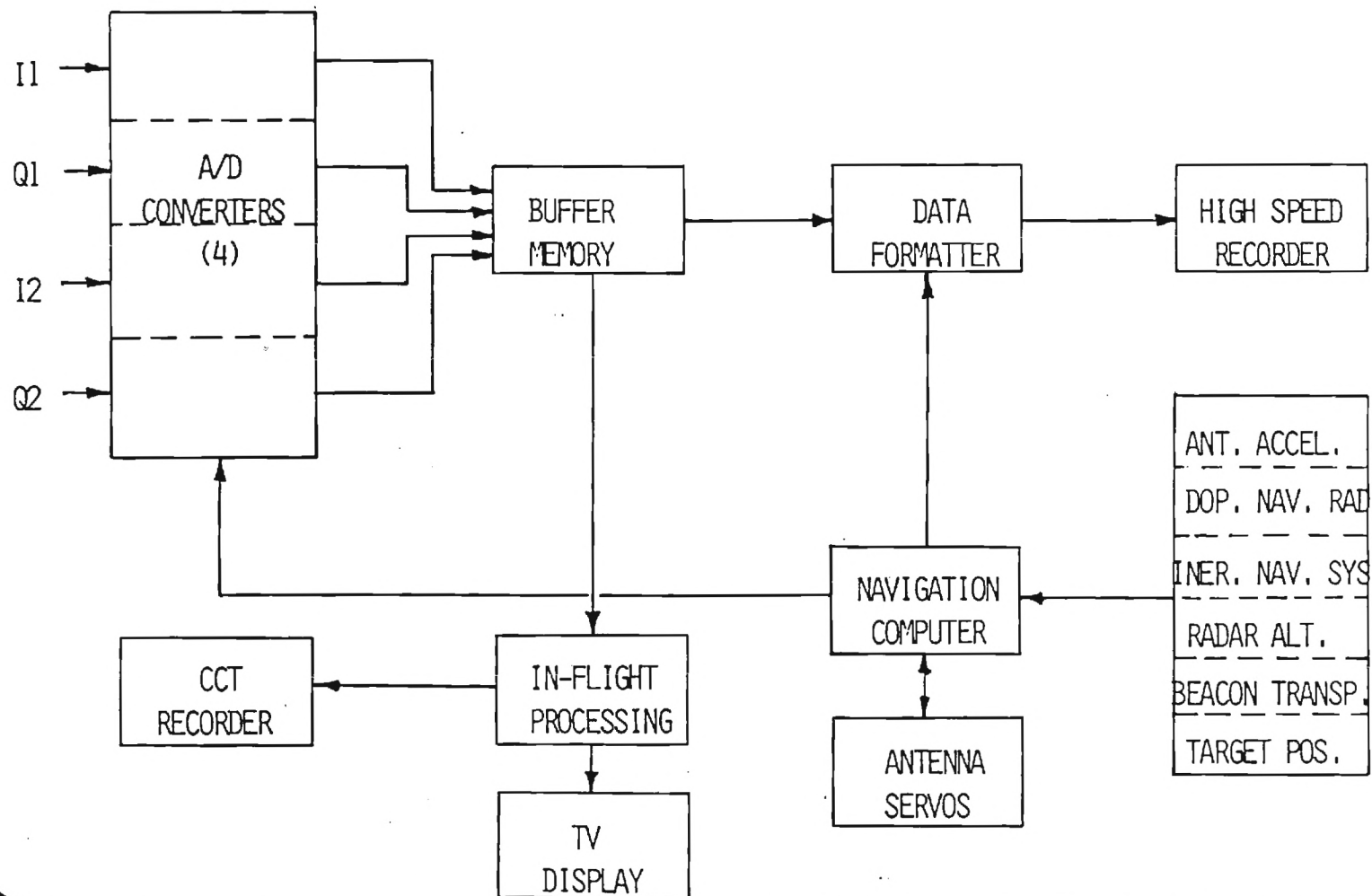
## APPROACH

1. Record digitized I and Q, range-gated, video returns for both receiver polarization channels.
2. Utilize 3-axis accelerometers, Doppler navigation radar, and inertial navigation systems to provide antenna position data to be used for motion compensation.
3. Utilize aircraft and target position data plus the beacon transponder return to provide proper antenna pointing directions.
4. Monitor transmitter power, receiver output levels, noise levels; provide for signal calibration.
5. Provide in-flight ability to assess data quality by displaying medium resolution radar imagery in "real time."



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## SIMPLIFIED BLOCK DIAGRAM







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## ANALOG-TO-DIGITAL DATA CONVERSION

- o 8-bit, 32 MHz A/D converters are off-the-shelf items.
- o Range gate delay will be provided by the navigation computer from stored target position data and calculated aircraft position data.
- o Four A/D converters are required for full polarization matrix data collection capability.
- o An oversampling of the range dimension data by a factor of 1-1/2 is provided.



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## BUFFER MEMORY

- o The data rate from the A/D converters is slowed down from a burst rate to a steady rate into the recorder formatter.
- o Buffering is performed on all four receiver channels simultaneously at the pulse repetition rate.



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## NAVIGATION SENSORS

### ANTENNA ACCELEROMETERS

3-axis system provides  $\ddot{x}$ ,  $\ddot{y}$ , and  $\ddot{z}$  on the gimbal.

### DOPPLER NAVIGATION RADAR

True ground speed is provided from  $\dot{x}$ ,  $\dot{y}$ , and  $\dot{z}$  of the aircraft.

### INERTIAL NAVIGATION SYSTEM

Latitude, longitude, and height plus roll, pitch, and yaw are provided

### RADAR ALTIMETER

Relative height of the aircraft to the local terrain below the aircraft is provided.

### BEACON TRANSPONDER

Specific target area locations are designated.

### TARGET POSITIONS

Stored target position information will be referenced to beacon transponder locations.



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## NAVIGATION COMPUTER

- o Provides range gate delay to the A/D converters for sampling initiation.
- o Calculates antenna pointing direction.
- o Commands the gimbal position.
- o Compares gimbal position to command position as part of built in test.



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## HIGH SPEED RECORDER

- o Duplicate of the ground-based recorder, except possibly for playback capability.
- o Georgia Tech requests that high speed recorders be Government furnished equipment.
- o Provides a digital tape that contains all sampled radar video signals, motion compensation data, time code data, polarization status flags, range bin position, etc.



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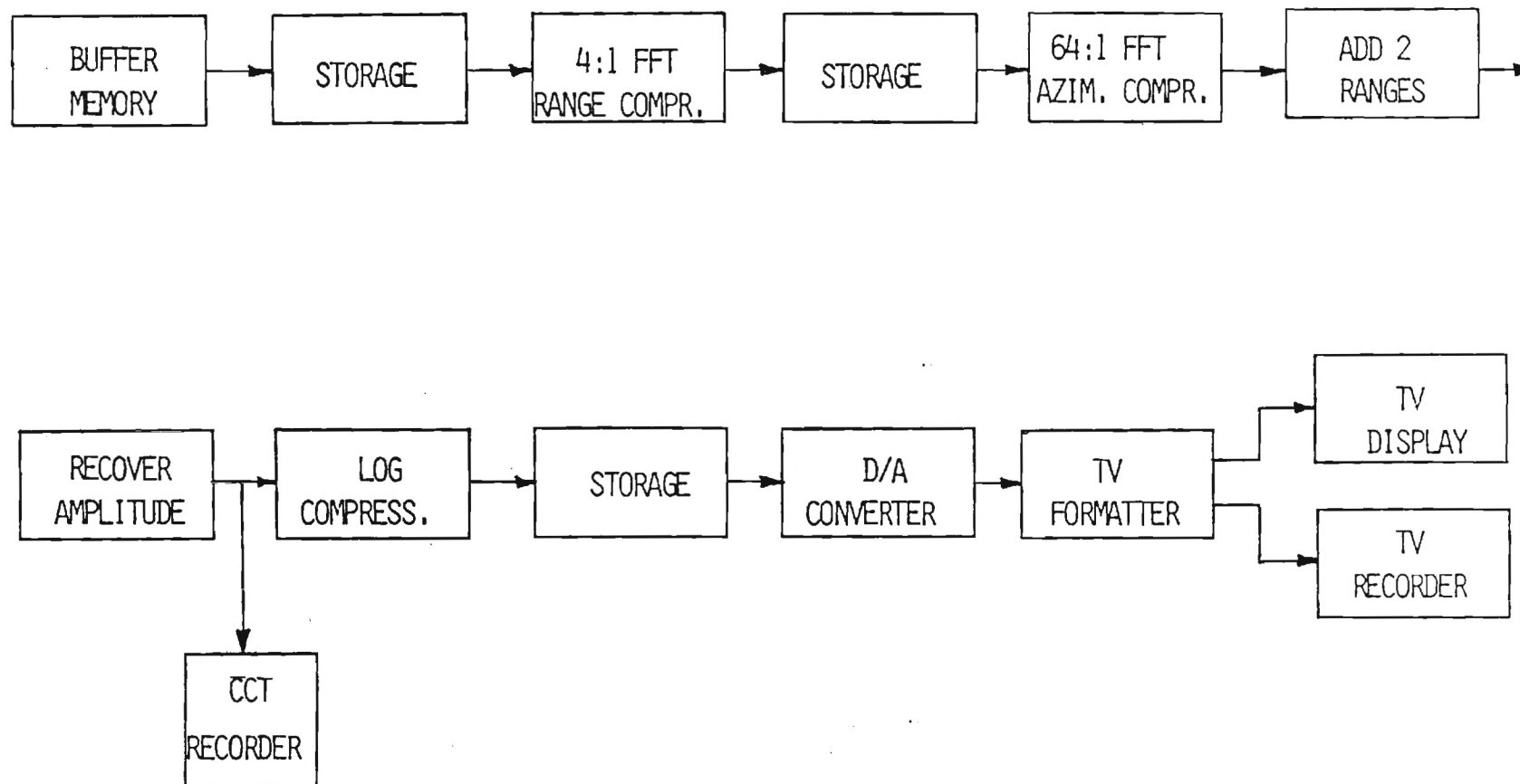
## IN-FLIGHT DATA PROCESSING OBJECTIVES

- o Processing provides an ability to assess the quality of the radar data in a "real time" manner.
- o Processing provides a means of documenting radar scenes in a television compatible format.



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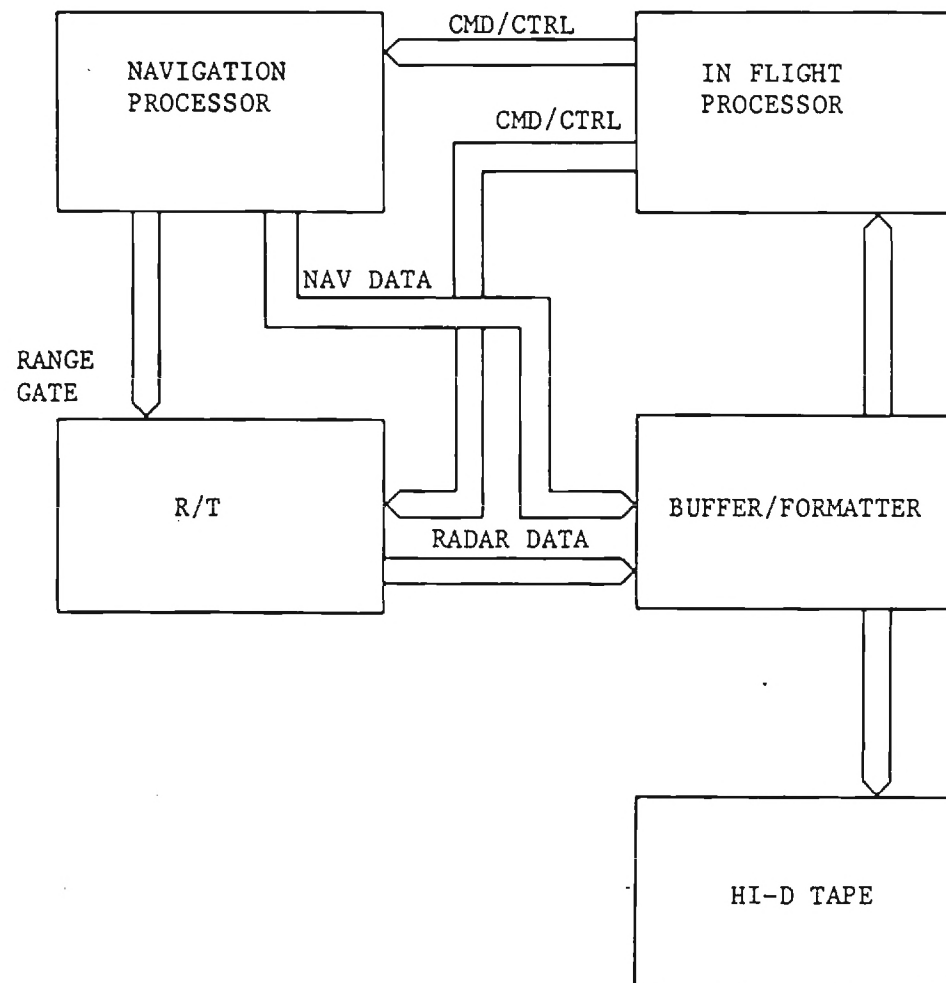
## IN-FLIGHT DATA PROCESSING





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## AIRBORNE DIGITAL SYSTEM - OVERVIEW

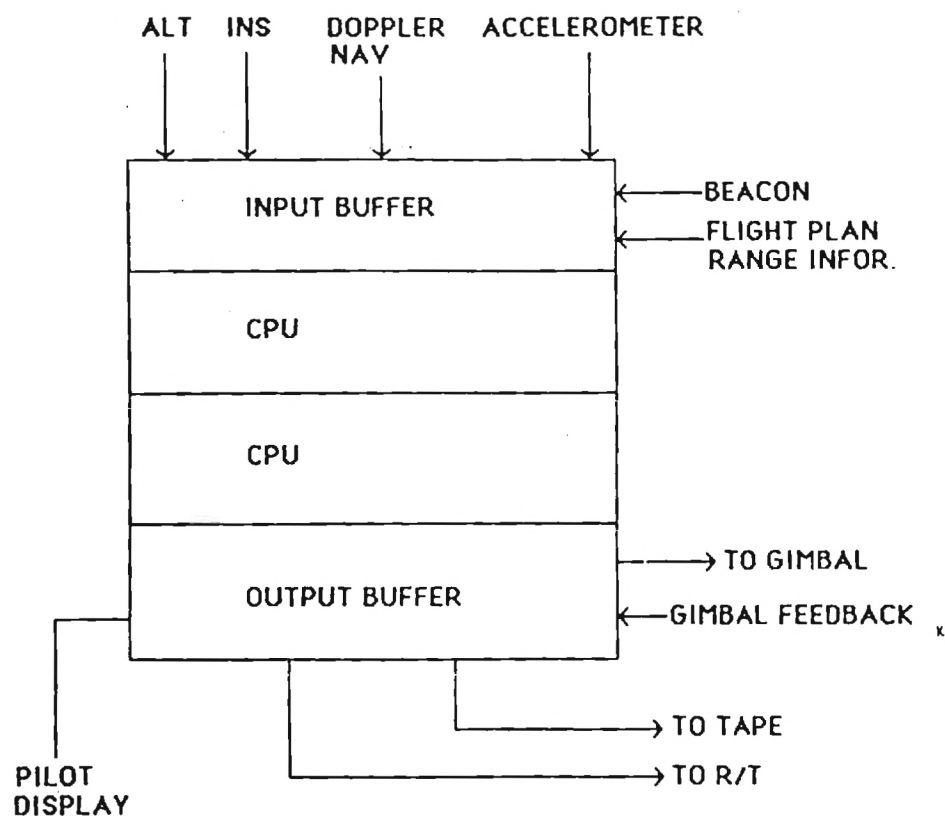






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## NAV AND STABILIZATION PROCESSOR



### RECORD FORMAT

INS  $\Lambda, \lambda, h, R, P, Y$

D.N  $\dot{X}, \dot{Y}, \dot{Z}$

ACCEL.  $\ddot{X}, \ddot{Y}, \ddot{Z}$

ALT. ALT

GIMBAL COMM. AZ, EL

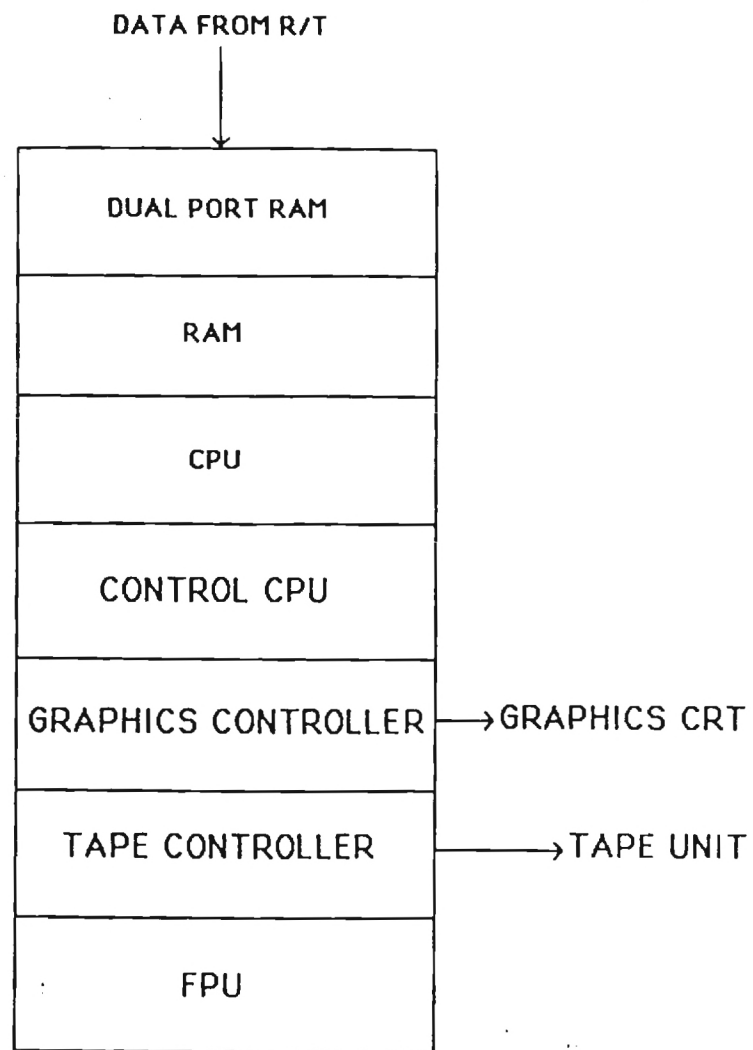
GIMBAL FEEDBACK AZ, EL

RANGE RATE DIGITAL COUNT



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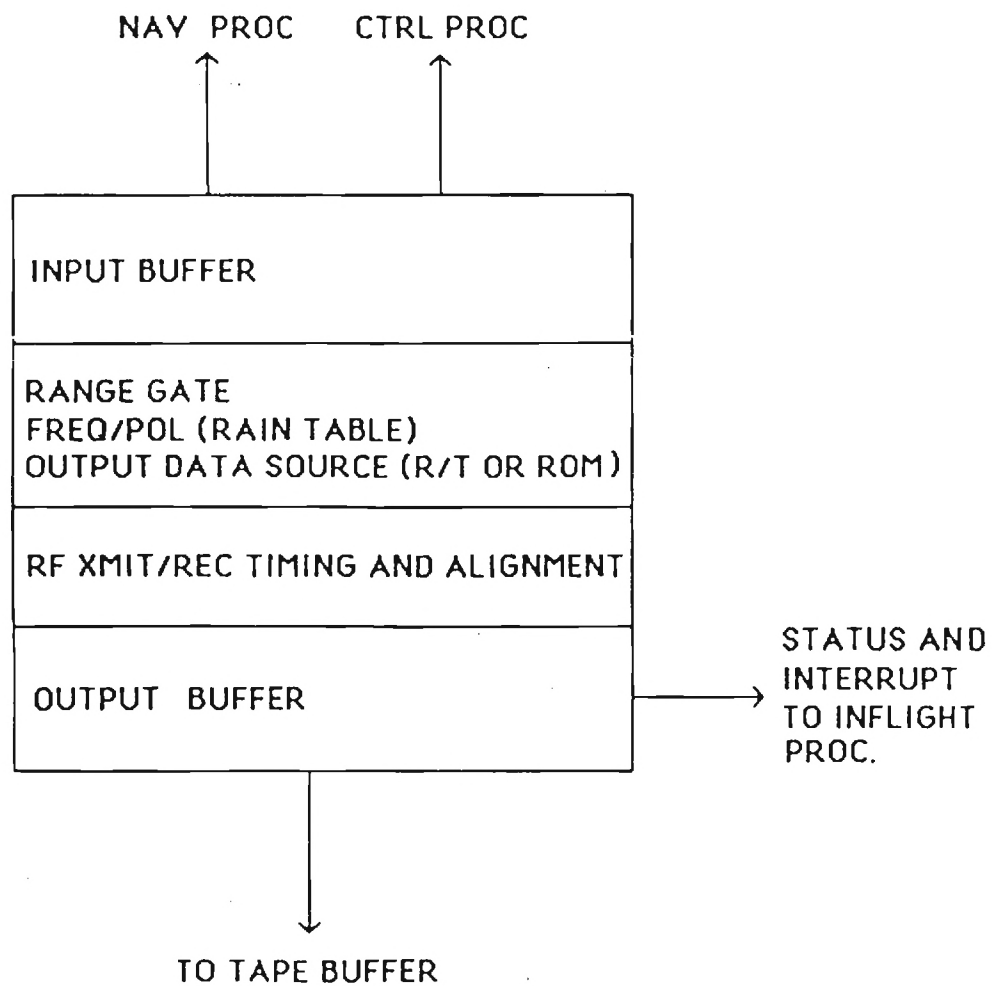
## IN FLIGHT DATA PROCESSOR





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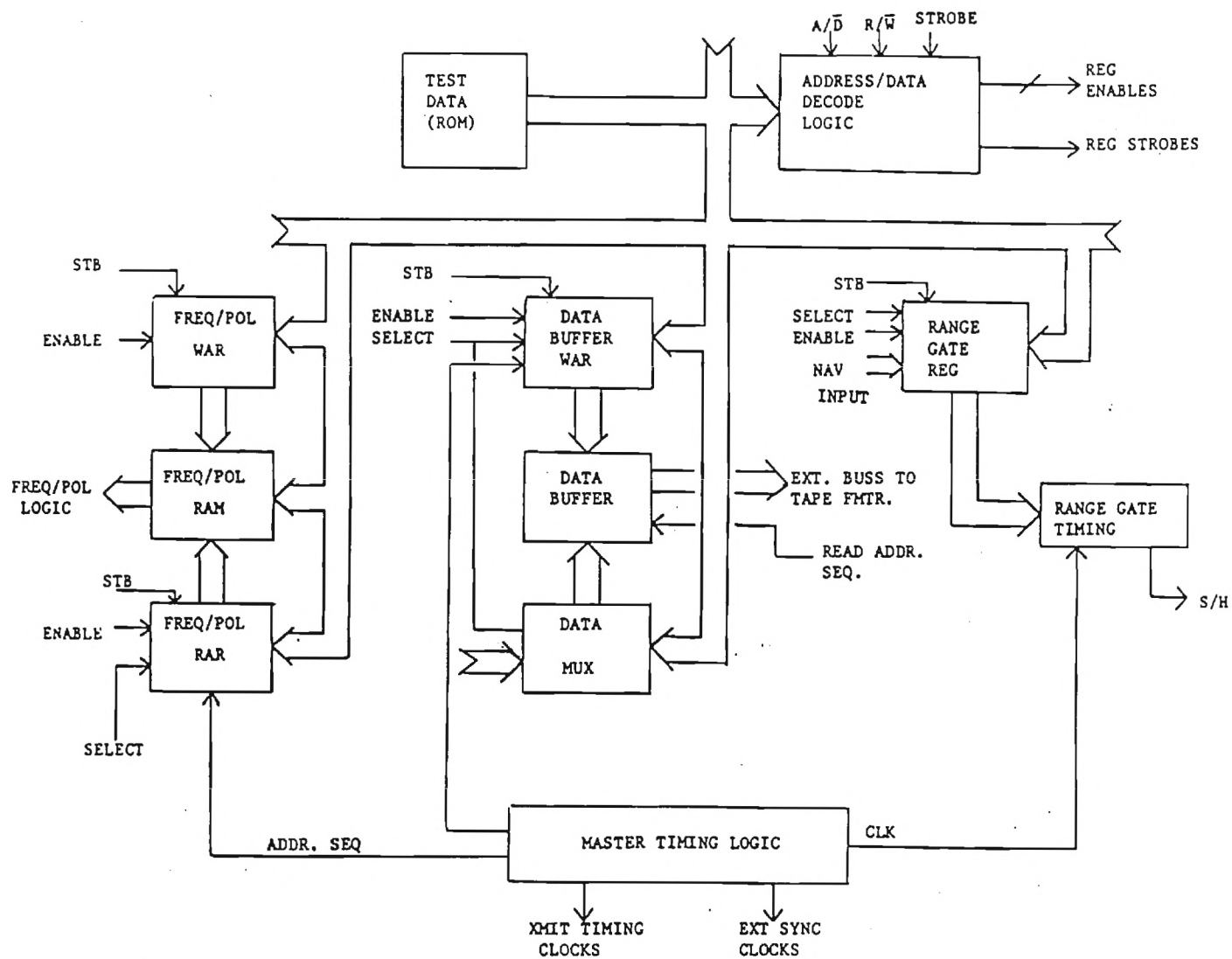
## R/T DIGITAL SUBSYSTEM





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## R/T DIGITAL SUBSYSTEM OVERVIEW





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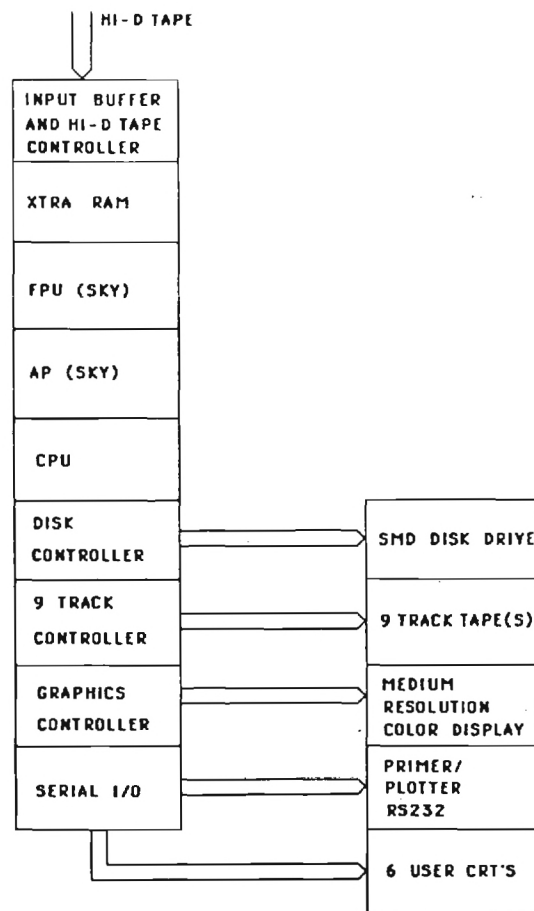
## GROUND BASED DATA ACQUISITION

- COMMERCIAL 32-BIT MICROPROCESSOR BASED TECHNOLOGY
- REAL TIME PROCESSING APPROACH
- QUICK LOOK PROCESSING
- DATA BASE PROCESSING
- INFLIGHT DEVELOPMENT SYSTEM
- PROCESSOR COMMONALTY
- COMPACT SYSTEM



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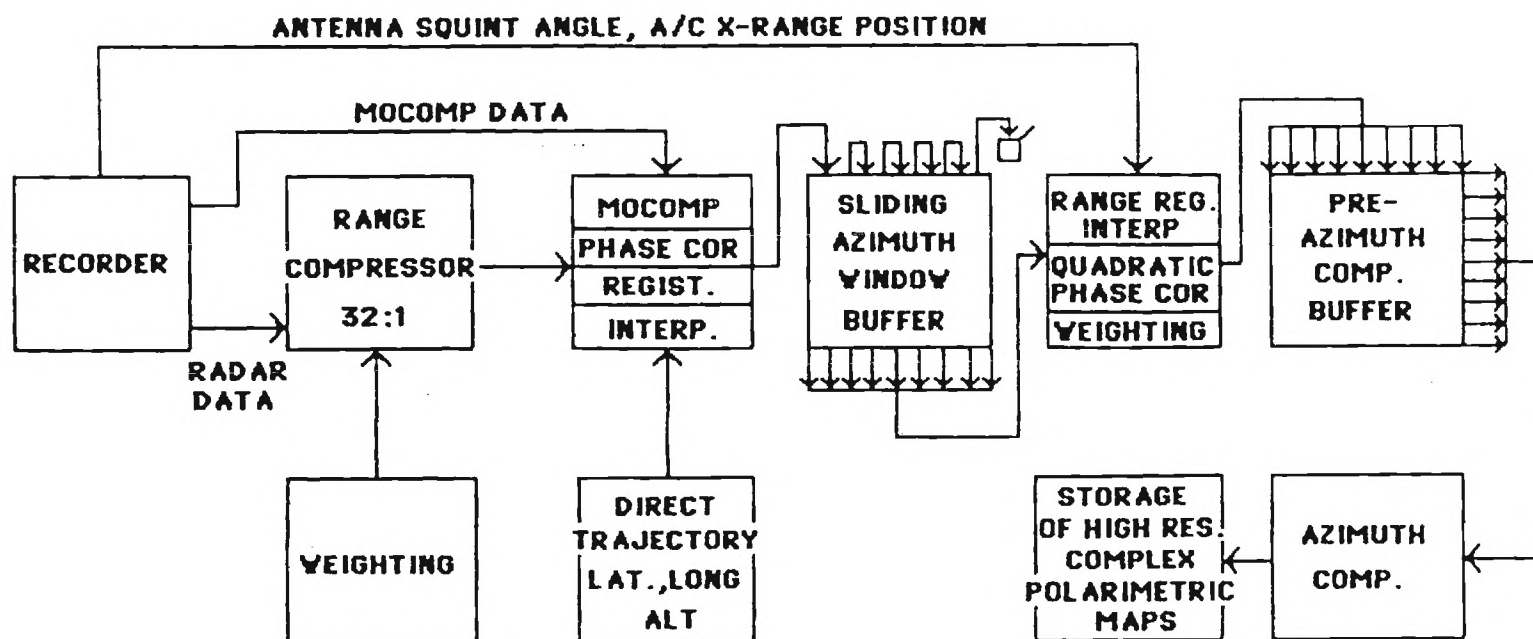
## GROUND BASED ANALYSIS SYSTEM





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## SAR GROUND PROCESSING STATION (STRIP MAP & SPOTLIGHT MODES)





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## GROUND BASED PROCESSING SUMMARY

- QUICK LOOK CALIBRATION VERIFICATION
- QUICK LOOK BITE ANALYSIS
  - GIMBAL POINTING
  - R/T TEST TARGET
  - R/T PERFORMANCE ANALYSIS
- QUICK LOOK SCENES
  - 11 MIN - SAR
  - 24 MIN - SPOTLIGHT
- CALIBRATED DATA BASE





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## SYSTEM CALIBRATION

### AMPLITUDE

- PRE AND POST FLIGHT CALIBRATIONS (GROUND)
- IN FLIGHT CALIBRATION
- CALIBRATION ASSESSMENT (QUICK LOOK)

### RESOLUTION

- IN FLIGHT CALIBRATION
- CALIBRATION ASSESSMENT



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## AMPLITUDE PRE AND POST FLIGHT CALIBRATION

- SIGNAL INJECTION (RECEIVER TRANSFER CURVE)
  - + INJECT TRANSMITTED SIGNAL INTO RECEIVER AT ALL FREQUENCIES
  - + VARY POWER LEVEL FROM RECEIVER SATURATION TO NOISE
  
- CALIBRATED TARGET STARE (ABSOLUTE CALIBRATION)
  - + STARE AT ONE OR MORE ODD BOUNCE TARGETS AND ONE OR MORE EVEN BOUNCE TARGETS
  - + PLACE TARGETS AT PROPER HEIGHT TO ELIMINATE MULTIPATH



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## IN FLIGHT CALIBRATION

- FLY OVER CALIBRATED TARGET ARRAY
  - + EVEN AND ODD BOUNCE TARGETS
  - + VARYING DISTANCES BETWEEN TARGETS
  - + OPTICAL MARKERS FOR TV VIEWING



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## CALIBRATION ASSESSMENT

- COMPARE INJECTED SIGNAL TRANSFER CURVE TO ABSOLUTE GROUND CALIBRATION USING RADAR EQUATION
- COMPARE IN FLIGHT ARRAY CALIBRATION TO GROUND CALIBRATION
- TWO INDEPENDENT CALIBRATION CHECKS



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## RESOLUTION CALIBRATION

### INFLIGHT CALIBRATION

- FLY OVER RADAR REFLECTOR ARRAY WITH VARYING DISTANCE BETWEEN REFLECTORS

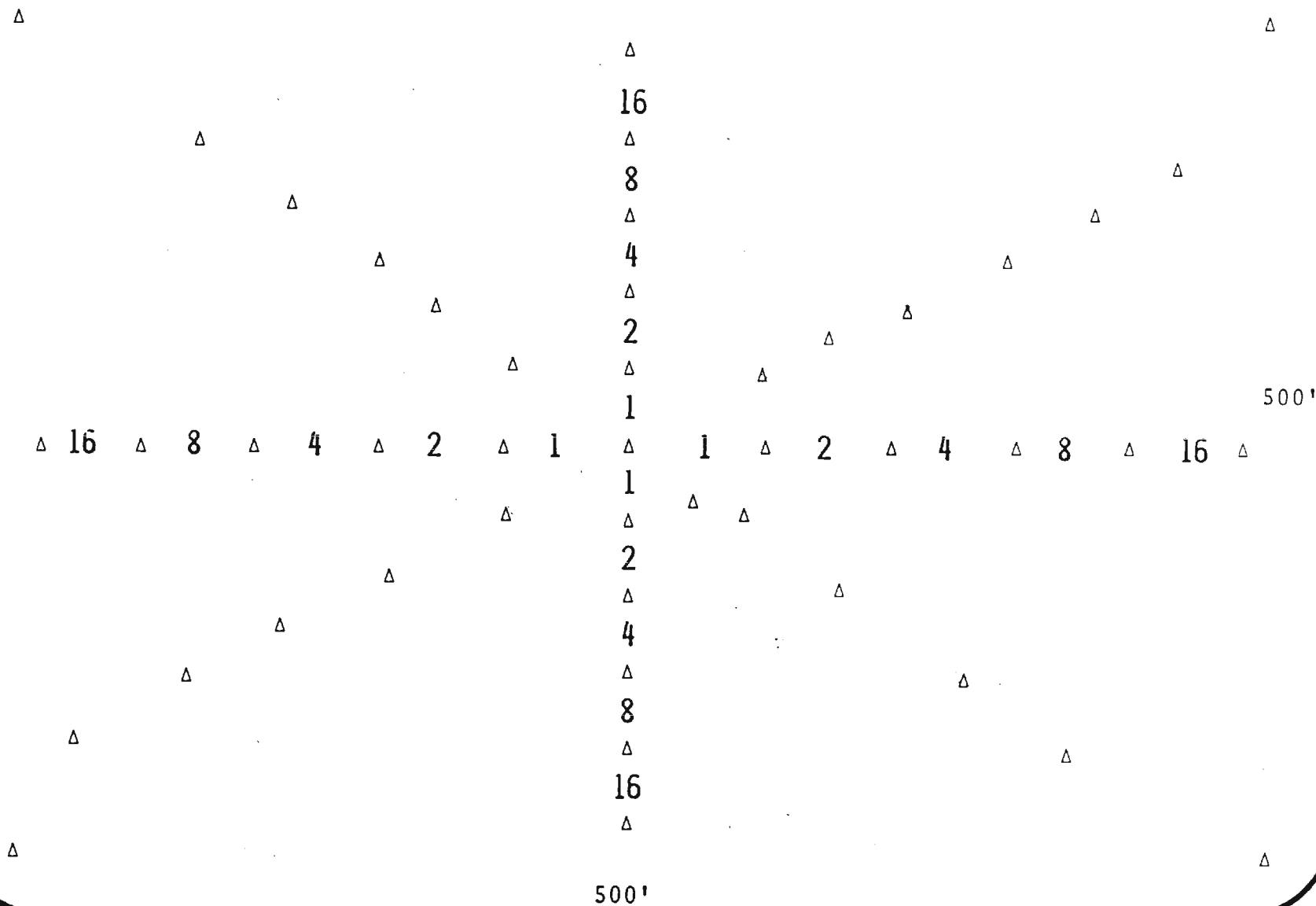
### RESOLUTION ASSESSMENT

- DISPLAY CALIBRATION DATA ON XY DISPLAY
- DETERMINE THE RESOLUABLE REFLECTOR DISTANCE



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## CALIBRATION REFLECTOR ARRAY





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## DAILY FLIGHT TEST PROCEDURES

- o SYSTEM CHECK OUT
  
- o GROUND CALIBRATION  
+ SIGNAL GENERATOR  
  
+ ABSOLUTE STARE
  
- o FLIGHT CALIBRATION (FLY OVER REFLECTOR ARRAY)
  
- o BEGIN FLIGHT TESTS  
VIEW MEDIUM RESOLUTION DISPLAY TO ASSESS DATA QUALITY BETWEEN RUNS
  
- o POST FLIGHT GROUND CALIBRATION



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## DAILY QUICK LOOK PROCEDURES

- O CLOSE SIGNAL GENERATOR AND ABSOLUTE GROUND CALIBRATIONS - COMPARE PRE- AND POST-FLIGHT CALIBRATIONS
- O PROCESS SELECTED DATA SCENES INCLUDING THE AIRBORNE CALIBRATION ARRAY
- O CHECK CALIBRATION ARRAY FOR CALIBRATION CLOSE WITH GROUND CALIBRATION AND FOR RESOLUTION
- O ENTER ASSESSMENT OF DATA QUALITY OF SAMPLED RUNS INTO DATA BASE
- O PRODUCE 35 MM IMAGES OF SCENES





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## DAILY PRODUCTION PROCEDURES

- 0 DETERMINE SCENES TO BE PROCESSED FROM PRIORITY LIST
- 0 LOAD CALIBRATION FILE FOR A SET OF RUNS
- 0 PROCESS SCENES
- 0 DUMP PROCESSED SCENES TO CCT
- 0 EVALUATE SCENES
- 0 COMPUTE STATISTICS/REGISTER WITH GROUND TRUTH
- 0 ENTER SUMMARIES INTO DATA BASE
- 0 TRANSCRIBE TO ARCHIVAL STORAGE MEDIUM
- 0 PRODUCE 35 MM IMAGES OF SCENES



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## DATA PROCESSING TRADE OFF

WITH THE PROPOSED SYSTEM A  $1024 \times 1024$  SCENE CAN BE PROCESSED IN APPROXIMATELY ELEVEN MINUTES TO ACHIEVE A CALIBRATED, COMPLEX, HIGH RESOLUTION DATA SET.

THE TIME TO PROCESS ONE  $1024 \times 1024$  SCENE IS THUS

$$11 \text{ MINUTES/SCENE} \times 4 \text{ POLARIZATIONS} = 44 \text{ MINUTES.}$$

THUS, THE MAXIMUM NUMBER OF  $1024 \times 1024$  SCENES THAT CAN BE PROCESSED IN AN 8 HOUR DAY IS

$$480 \text{ MINUTES/DAY} / 44 \text{ MINUTES PER SCENE} = 11 \text{ SCENES/DAY}$$

THUS THE NUMBER OF SCENES THAT CAN BE PROCESSED IN THE TWO YEAR TEST PERIOD IS

$$104 \text{ WEEKS} \times 5 \text{ DAYS/WEEK} \times 11 \text{ SCENES/DAY} = +5720 = 24.0 \text{ GIGA PIXALS}$$

IN PRACTICE IT WOULD BE IMPOSSIBLE TO DO MORE THAN GLANCE AT SUCH A LARGE NUMBER OF SCENES. THUS, A COLOR DISPLAY WILL BE USED AS A BASIC ANALYSIS TOOL TO ASSESS THE SCENE INTEGRITY. IN THIS MANNER THE SCENES CAN BE PROCESSED AND VIEWED QUICKLY AND ONLY USEFUL SCENES SELECTED FOR PERMANENT STORAGE AND LOGGING IN THE DATA BASE. THUS, VIRTUALLY ALL OF THE DATA COULD BE LOOKED AT AND A SUBSET SELECTED FOR ADDITIONAL ANALYSIS. AS AN OPTION, AN EXTRA ARRAY PROCESSOR COULD DOUBLE THE SCENES THAT COULD BE PROCESSED



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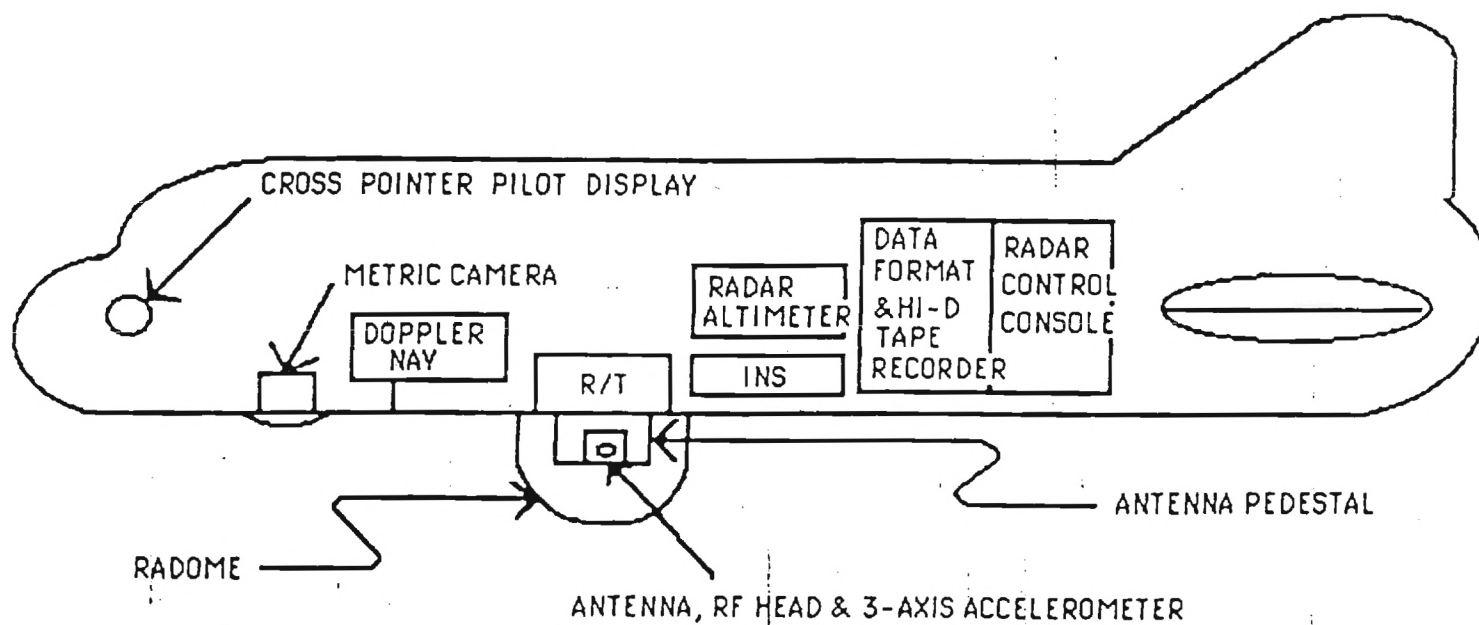
## C-131B SPECIFICATIONS

WING SPAN	106	FT.
LENGTH	79	FT.
HEIGHT	28	FT.
EMPTY WEIGHT	32,000	LBS.
GROSS WEIGHT	53,200	LBS.
MAXIMUM SPEED	295	KNOTS
CRUISING SPEED	220	KNOTS
INITIAL CLIMB	1,230	FT./MIN.
APPROXIMATE MAXIMUM TEST ALTITUDE	20,000	FT. MSL.
RANGE	1,300	NMI
RADOME POSITION	MUST BE ABLE TO ELEVATE ANTENNA 2° WITH UNOBSTRUCTED VIEW ON THE GROUND	



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## CONVAIR CV240 (T29) OR CONVAIR CV340 (C131)





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## ESTIMATED COST SUMMARY TOTAL

<u>ITEM</u>	<u>SALARY COSTS</u>	<u>MATERIALS</u>	<u>TRAVEL</u>	<u>TOTAL</u>
MANAGEMENT	451,371	20,000	30,000	510,350
SYSTEM ENGINEERING	350,366	1,000		351,366
SYSTEM INTEGRATION	365,636	26,400		392,036
GYMBAL	244,467	330,000	4,000	578,467
RADOME	61,395	100,000		161,395
TRANSMITTER	100,915	10,000	1,609	112,525
RF, IF, VIDEO, DATA ACQUISITION, RADAR TIMING AND CONTROL	766,686	279,500	5,000	1,051,186
PEDESTAL	200,544	350,000		550,544
DATA ACQUISITION - NAV SYSTEM	116,511	20,000	5,000	141,511
DATA ACQUISITION - R/T DIGITAL	174,720	25,000		199,720
DATA ACQUISITION - INFLIGHT PROCESSOR	117,190	35,000		152,190
DATA ACQUISITION - INTERCONNECT	76,520			76,520
DATA ACQUISITION - SYSTEM DESIGN	129,417		5,000	134,417
DATA ACQUISITION - DATA ANALYSIS VAN	300,991	80,000	5,000	385,991

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TOTAL PREPARATION COST 4,790,238



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## ESTIMATED COST SUMMARY TOTAL (CONTINUED)

	<u>SALARY COSTS</u>	<u>MATERIALS</u>	<u>TRAVEL</u>	<u>TOTAL</u>
FLIGHT TEST SUPPORT	642,867	100,000	120,000	862,867
QUICK LOOK ANALYSIS	287,553	50,000	80,000	417,553
PRODUCTION IMAGE PROCESSING	350,000	160,000	10,000	520,000

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TOTAL TEST/ANALYSIS COST (2 YEARS)	1,800,420
COST PER SCENE (10,000 SCENES)	180
TOTAL COST	6,590,658

### OTHER ITEMS (ASSUMED GFE)

### TOTAL

AIRPLANE	
INSTALLATION	558,427
TEST SUPPORT	2,164,520
INS & BEACON	200,000
HIGH DENSITY TAPE RECORDER	200,000
DOPPLER NAVIGATION SYSTEM	75,000

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TOTAL GEF ITEMS	3,197,947
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## RECENT GTRI HARDWARE/MEASUREMENT PROGRAMS

<u>SPONSOR</u>	<u>DESCRIPTION</u>	<u>START DATE</u>	<u>DURATION</u>	<u>ACTUAL AMOUNT</u>	<u>% OF ESTIMATE OVERUN/ (UNDERRUN)</u>
ERADCOM	95 GHZ MODULATOR	6/81	7 mos.	\$99,000	0%
GULF INTERSTATE CORPORATION	LOW FREQUENCY GROUND PENETRATION RADAR	9/81	21 mos.	\$500,000	0%
TELEDYNE BROWN CORP.	COHERENT REPEATER	10/81	9 mos.	\$290,000	0%
WESTINGHOUSE CORPORATION	HI RES 35 GHZ RADAR MEASUREMENTS	3/82	5 mos.	\$410,000	0%
MICOM	SNOWMAN MMW MEASUREMENTS	7/83	14 mos.	\$2,400,000	0%
WSMR	AN/MPS-36 RADAR MODIFICATION	9/82	24 mos.	\$1,123,000	5%



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## SENSOR SPECIFICATIONS

- A. THE PROCESSED SIGNAL-TO-NOISE RATIO SHALL BE 60 dB AT 1.5 KM FOR A 10 dBsm TARGET. THE SIGNAL-TO-CLUTTER RATIO WILL BE 33 dB FOR  $-13 \text{ dB } \sigma^0$ .
- B. THE SPECIFIED RANGE AND CROSS-RANGE RESOLUTION SHALL BE OBTAINED IN BOTH THE STRIP MAP AND SPOTLIGHT SAR MODES.
- C. ALL RADAR AND NAVIGATION SIGNALS WILL BE RECORDED IN RAW FORM TO SUPPORT SAR POST-PROCESSING.
- D. THE ABSOLUTE SETTTLING TIME OF THE SYNTHESIZER PROPOSED IS  $1 \mu\text{s}$  WHICH FOR THE PRF RATE OF 32 KHz YIELDS A MAXIMUM RANGE OF 4.5 KM.
- E. POLARIZATION ISOLATION SHALL BE 25 dB OVER THE MAIN BEAM GIVEN THE COMBINED EFFECTS OF ANTENNA AND RADOME.





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## SENSOR SPECIFICATION (CONTINUED)

- F. THE STABILITIES REQUIRED FOR THE DESIRED SIDELobe PERFORMANCE OVER THE 0.25 SECOND INTEGRATION PERIOD ARE SATISFIED.
- G. THE CROSS-RANGE SWATH SHALL BE 150 M IN THE SPOTLIGHT MODE AND UP TO 1500 M IN THE STRIP MODE.

THE RANGE SWATH COVERAGE SHALL BE 100 M AT  $50^{\circ}$  AND 150 M AT  $10^{\circ}$  DEPRESSION ANGLE.

- H. THE TARGET LOCATION ERROR SHALL NOT EXCEED 10 FT. RANGE AND CROSS RANGE.